



# MAINTENANCE PROGRAM AIRBUS HELICOPTERS EC 130 T2

## Appendix D06 – ALS Inspection 600 Hours

Reg. Mark : PK - Date : \_\_\_\_\_  
MSN : \_\_\_\_\_ Station : \_\_\_\_\_  
TSN / CSN : \_\_\_\_\_ WO No. : \_\_\_\_\_

ITEM CODE NO.	CHAPTER	TASK	ENGINEER'S SIGNATURE, STAMP & DATE
55/11/00 /000/000 /020	55-11	<b>Tail boom / Fenestron junction frame</b> POST MOD 074775 & PRE MOD 074581 Visual check for absence of cracks. GVI <b>AMM 55-11-00, 6-4</b>	
62/11/00 /000/000 /045	62-11	<b>Main rotor blade</b> 355A11-0030-04 (-) Monitoring of the stainless steel shielding adhesion using tapping test. NOTE: On new blade or after replacement of one or both stainless steel shieldings, the new stainless steel shieldings are covered by a shorter interval, every 150 FH during the first 450 FH. DI <b>AMM 62-11-00, 6-1</b>	
65/11/00 /000/000 /030	65-11	<b>Bearing</b> 593404 (704A33651181) Check and greasing bearing individually after disconnection bearing block. DI LUB <b>AMM 65-11-00, 3-1</b> <b>AMM 65-11-00, 6-5</b>	
*** End of Appendix D06 Items ***			

PERSONNEL PRTICIPATING IN THIS INSPECTION			
NAME	POSITION	SIGNATURE	LICENSE NUMBER

### RETURN TO SERVICE

The work recorded above has been carried out in accordance with the requirements of the Civil Aviation Safety Regulation for the time being in force and in that respect the aircraft is consider fit for Release to Service.

Name : \_\_\_\_\_ Stamp : \_\_\_\_\_  
Signature : \_\_\_\_\_ Place/Date : \_\_\_\_\_