



**AIRCRAFT CHECK WORK SUMMARY**  
(Form: SCA/MTC/051)

DATE OF ISSUED	JO/WO #	TYPE OF MAINTENANCE	DATE OF ACCOMPLISHED		
28 Sept 2022	WO/022-SNI/IX/2022	Inspection Doc.15,01, 100 FH			
A/C Type	Mfg. Serial Number	A/C Registration			
C208B	C208B-5068	PK-SNI			
<b>AIRCRAFT DATA</b>					
Subject	Pos #	Serial Number (SN)	TTSN/TCSN		
Engine	#1	PCE-VA0073			
	#2	-			
Propeller/Rotor	#1	200707			
	#2	-			
Landing Gear	NLG				
	LH MLG				
	RH MLG				
<b>PACKAGE COVERED</b>					
No	Subject	Qty	Remark		
1	Non-Routine Card	-			
2	Inspection Card	1			
3	Work Order	1			
4	Summary Inspection List	1			
5	Material and Tool List	-			
6	Escalation form	-			
7	CRS (SMI / Unscheduled Maintenance)	1			
<b>INSPECTION CARD (IC) LIST (Finding during maintenance)</b>					
No	Taskcard Ref	Subject	Status		Name/ Sign & Stamp
			Open	Close	
<u>IC-001</u>					
<u>IC-002</u>					
<u>IC-003</u>					
<u>IC-004</u>					
<u>IC-005</u>					
<u>IC-006</u>					

<u>IC-007</u>					
<u>IC-008</u>					
<u>IC-009</u>					
<u>IC-010</u>					
<u>IC-011</u>					
<u>IC-012</u>					
<u>IC-013</u>					
<u>IC-014</u>					
<u>IC-015</u>					

Prepared by :  
Technical Support


Checked by :  
Chief Maintenance

Verified by :  
Chief Inspector

Approved by :  
Technical Manager



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# SUMMARY INSPECTION ITEMS (Form: SCA/MTC/050)

WO Ref: WO/022-SNI/IX/2022

NO.	TASK CARD NO.	DESCRIPTION	DATE	EST MHR	NAME	STAMP
1	B03	PT6A-140 ENGINE GROUND RUN PERFORMANCE				
2	CHAPTER 51	ENGINE PT6A-140 100 HOUR INSPECTION/ MINOR INSPECTION				
3	CHAPTER 07	INSPECTION DOCUMENT 01 – 12 MONTHS				
4	CHAPTER 21	INSPECTION DOCUMENT 15 – 7500FH INITIAL, 2500FH REPETITIVE (SID)				
5	SCA/MTC/0 23	EMERGENCY EQUIPMENT CHECK				



PT. SMART CAKRAWALA AVIATION

## CERTIFICATE RETURN TO SERVICE

SCHEDULED MAINTENANCE INSPECTION  
(CRS-SMI)

A/C TYPE : CESSNA 208B

TTSN :

A/C REG : PK-SNI

TCSN :

MSN : C208B-5068

DATE :

TYPE OF INSPECTION : INSPECTION DOC.15, DOC.01 & 100 HOURS

DUE AT : 7476.0 HOURS / OCT 2022

REF : MP C208B REV.12

EXCEPTION

### AUTHORIZED PERSON


I hereby certify that this aircraft has been maintained accordance with CASR and Maintenance Program.  
Aircraft safe and airworthy for flight

NAME	CAT	AMEL/OTR NO	SIGN&STAMP	DATE
	AIRFRAME & POWER PLANT			
	EIRA			

THE NEXT DUE TYPE OF INSPECTION :

DUE AT :

Form: SCA/MTC/049

	<b>INSPECTION CARD</b> <b>(Form: SCA/MTC/ 048)</b>	TECHNICAL DEPARTMENT
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1. CARD #	2. JO/WO #	3. ORIGINATOR	4. CARD REF	5. DATE
6. A/C REG/MSN	7. A/C TYPE	8. TRADE	12. VENDOR ORDER #	
9. ZONE	10. STA	11. MTC TYPE		

13. DESCRIPTION/DEFECT-IF FINDING OF CPCP INSPECTION, PLEASE COMPLETE SET. 20	14 PPC/ENG	15 DATE

16. CORRECTIVE ACTION	17 MECH	18 ENG. LIC	19 DATE
Performed at A/C TT : ..... A/C TC /LDG : .....			

20. CORROSION INFORMATION					
LOCATION	CAUSE OF DAMAGE				
	<input type="checkbox"/> Environment				
	<input type="checkbox"/> Internal Leakage				
CORROSION <input type="checkbox"/> Isolated <input type="checkbox"/> Widespread	<input type="checkbox"/> Chemical Spill				
CORROSION LVL <input type="checkbox"/> 1 <input type="checkbox"/> 2 <input type="checkbox"/> 3	<input type="checkbox"/> LAV/Galley Spill				
PROPOSED ACTION <input type="checkbox"/> Doublers	<input type="checkbox"/> Blocked Drain				
<input type="checkbox"/> Others	<input type="checkbox"/> Wet Insulation Blanket				
	<input type="checkbox"/> Other				

21. If the defect is RII, Please Sign this card finally by RII Inspector	INSP	DATE
NOTICE OF INSPECTOR		

22. PARTS REQUIRED						
PART DESCRIPTION	PART NO	QTY	SERIAL NO		STATUS	
			ON	OFF	CLOSE	OPEN

23. TOOLS REQUIRED			
DESCRIPTION	PART NO. / MODEL	NEXT CALIBRATION DATE	STATUS



# MAINTENANCE PROGRAM CESSNA C208/C208B

## Appendix B03 – PT6A-140 Engine Run Performance Sheet

Reg. Mark : PK - SNI

WO/FML No. : WO/022-SNI/IX/2022

PRE – INSPECTION	
Location	
Date	
Cycle	
Filed Barometric	
OAT	
Altitude	

POST – INSPECTION	
Location	
Date	
Cycle	
Filed Barometric	
OAT	
Altitude	

PRE – INSPECTION		
	Target	Actual
Tq		
Np		
ITT	°C	°C
Ng	%	%
Wf		
Oil Press		°C
Oil Temp		°C
Start Temp		°C

POST – INSPECTION		
	Target	Actual
Tq		
Np		
ITT	°C	°C
Ng	%	%
Wf		
Oil Press		°C
Oil Temp		°C
Start Temp		°C

Engine Run Up Checks					
Inertial <input type="checkbox"/>	EPL <input type="checkbox"/>	OVG <input type="checkbox"/>	Stby Alt <input type="checkbox"/>	BOV <input type="checkbox"/>	Brake <input type="checkbox"/> Randomn <input type="checkbox"/>
<b>NOTE:</b> 1. Brake system at Torque 2000 ft-lbs.      3. EPL check can't exceed 4% Ng per second.      5. Low idle at 55.5 - 57% 40Amps. 2. Inertial Separator at Torque 400 ft-lbs.      4. Standby Alt at 80% Ng.      6. High idle at 64 - 66% Ng 40Amps					

Engine Performance Target Table (Cessna C208B EX)

OAT (°C)	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41
Tq (ft.lbs)	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397
Np	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900
ITT (°C)	835	837	839	841	841	841	841	841	841	842	843	844	846	846	846
Ng (%)	102.7	102.7	102.7	102.7	102.7	102.7	102.7	102.6	102.6	102.6	102.6	102.6	102.6	102.6	102.5
WF (PPH)	578	578	578	578	578	578	578	570	565	565	560	560	555	548	548

**Note:**

1. Make sure that inertial separator in normal condition, no bleed air extracted from the engine and air condition OFF.
2. This table only applies to altitude 0-500 feet MSL. For higher altitude, refer to EMM 72-00-00.
3. Max fuel flow is 580 lb/hr fuel flow is not more than 15 lbs/hr higher than the value shown in table.
4. If parameters are outside the target performance table to EMM chapter 71-00-00.

REMARKS:
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PERFORMED BY			
Name	Sign & Stamp	Date	Location



# MAINTENANCE PROGRAM CESSNA C208/C208B

## Chapter 51 – Engine PT6A-140 100 Hours/Minor Inspection

Reg. Mark	:	PK - SNI	Date	:	
MSN	:	208B-5068	Station	:	
TSN / CSN	:		WO No.	:	WO/022-SNI/IX/2022

ITEM CODE NO.	ZONE	TASK	SIGNATURE	
			SIGN	STAMP
F710001	130	Do a check of the FCU manual override system for static operation. For the engines installed with a manual override system only.		
F710003	130	Do a compressor performance recovery wash		
F720000	130	Do a visual inspection of the Control Linkages and wiring.		
F720001	130	Do a visual inspection of the engine exhaust duct welds.		
F720002	130	Do a visual inspection of the engine exhaust duct for cracks.		
F720003	130	Do a visual inspection of the gas generator case and the center fireseal.		
F720004	130	Do a visual inspection of the rear fireseal mounting ring.		
F722001	130	Do a visual inspection of the air inlet screen.		
F723000	130	Do a visual inspection with a mirror or a borescope inspection of the First-stage Compressor Rotor and the inlet case for corrosion		
F725005	130	Do a detailed inspection of the turbine exhaust duct.		
F731002	130	Do a check for the fuel pump installation and leaks.		
F731003	130	Do a check of the oil-to- fuel heater installation		
F731035	130	Do a visual Inspection of the Fuel - Oil Heat Exchanger Fuel Filter Element  <b>(CLEANING / REPLACEMENT)</b> P/N OFF: _____ P/N ON: _____		



### Chapter 51 – Engine PT6A-140 100 Hours/Minor Inspection

ITEM CODE NO.	ZONE	TASK	SIGNATURE	
			SIGN	STAMP
F731006	130	Do a check of the drain valve for installation and leaks		
		NOTE: There is no need to remove the drain valve if there is no leaks.		
F731007	130	Do a check of the flow divider for installation and leaks.		
F731008	130	Do a visual inspection of the P3 filter and drain valve.		
F731015	130	Do a visual inspection of Fuel Pump outlet filter.  <b>(CLEANING / REPLACEMENT)</b> P/N OFF: _____ P/N ON: _____		
F731018	130	Clean or replace the P3 filter based on condition, service experience or environment. <b>Note:</b> If corrossions are found, replace filter.		
F732001	130	Do a check of the FCU for installation, linkages and pneumatic tubes.		
F792000	130	Inspect and clean oil filter for debris.		

PERSONNEL PARTICIPATING IN THIS INSPECTION			
NAME	POSITION	SIGNATURE	LICENSE NUMBER

#### RETURN TO SERVICE

The work recorded above has been carried out in accordance with the requirements of the Civil Aviation Safety Regulation for the time being in force and in that respect the aircraft is consider fit for Release to Service.

Name : \_\_\_\_\_ Stamp : \_\_\_\_\_

Signature : \_\_\_\_\_ Place/Date : \_\_\_\_\_



# MAINTENANCE PROGRAM CESSNA C208/C208B

## Chapter 7 – Inspection Document 01

Reg. Mark	:	PK - SNI	Date	:	
MSN	:	208B-5068	Station	:	
TSN / CSN	:		WO No.	:	WO/022-SNI/IX/2022

ITEM CODE NO.	ZONE	TASK	SIGNATURE	
			SIGN	STAMP
C122106	801 802 803 804 921 922 923 924	Key Lock Lubrication Task 12-21-07-640		
A251000	801 802	Smoke Goggle General Visual Inspection Task 25-10-00-210		
B262001	215 216 251 252	Portable Fire Extinguisher Functional Check (Weight Check) Task 26-20-00-720		
B272003	211 212 213 214 217 218 233 234 253 254 257 258 311 312 320 341	Rudder System Functional Check (Float Kit Installation) Task 27-20-00-721		
B313101	312	Flight Data Recorder System Functional Check Task 31-31-00-720		
A321001	721 722	Main Landing Gear Detailed Inspection Task 32-10-00-220		
A322002	701	Drag Link Forward Support Seal General Visual Inspection (Airplanes 20800553 and On and 208B5076 and on) Task 32-20-00-210		
A324001	721 722	Brakes Detailed Inspection Task 32-40-00-220		
A324005	721 722	Main Landing Gear Wheels and Tires Detailed Inspection Task 32-40-00-222		
A324009	710	Nose Landing Gear Wheel and Tire Detailed Inspection Task 32-40-00-224		
B350101	231 232 251 252 255 256 311 312 801 802	Oxygen System Operational Check Task 35-01-00-710		



# MAINTENANCE PROGRAM CESSNA C208/C208B

## Chapter 7 – Inspection Document 01

ITEM CODE NO.	ZONE	TASK	SIGNATURE	
			SIGN	STAMP
A714101	130	Engine Wash Ring, Air Plenum, and Thermocouple (T1) Detailed Inspection Task 71-41-00-220		
*** End of Inspection Document 01 Items ***				

PERSONNEL PARTICIPATING IN THIS INSPECTION			
NAME	POSITION	SIGNATURE	LICENSE NUMBER

### RETURN TO SERVICE

The work recorded above has been carried out in accordance with the requirements of the Civil Aviation Safety Regulation for the time being in force and in that respect the aircraft is consider fit for Release to Service.

Name : \_\_\_\_\_ Stamp : \_\_\_\_\_  
Signature : \_\_\_\_\_ Place/Date : \_\_\_\_\_



# MAINTENANCE PROGRAM CESSNA C208/C208B

## Chapter 21 – Inspection Document 15

Reg. Mark	:	PK - SNI	Date	:	
MSN	:	208B-5068	Station	:	
TSN / CSN	:		WO No.	:	WO/022-SNI/IX/2022

ITEM CODE NO.	ZONE	TASK	SIGNATURE	
			SIGN	STAMP
A531009	231 232 233 234 251 252 253 254 255 256 257 258	Seat Rails and Attachment Structure Detailed Inspection (SID 53-10-07) Task 53-25-00-220		
A532003	255 256 257 258 803 804	Cargo and Passenger Door Doublers Special Detailed Inspection (SID 53-20-01) Task 53-10-00-251		
A532013	233 234	Bulkheads and Stiffeners Below the Seat Rail Attachments at FS 143.00 and FS 158.00 Detailed Inspection (SID 53-20-12) Task 53-25-00-221		
*** End of Inspection Document 15 Items ***				

PERSONNEL PARTICIPATING IN THIS INSPECTION			
NAME	POSITION	SIGNATURE	LICENSE NUMBER

### RETURN TO SERVICE

The work recorded above has been carried out in accordance with the requirements of the Civil Aviation Safety Regulation for the time being in force and in that respect the aircraft is consider fit for Release to Service.

Name	:	_____	Stamp	:	_____
Signature	:	_____	Place/Date	:	_____

END OF TASK

## **TASK 53-25-00-221**

### **3. Bulkheads and Stiffeners Below the Seat Rail Attachments at FS 143.00 and FS 158.00 Detailed Inspection**

#### **A. General**

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Retain printed data for historical reference only. For future maintenance, use only current data.

Page 1 of 6

Print Date: Thu Jan 06 20:20:35 CST 2022

- (1) This task includes the Supplemental Inspection Document (SID) requirements necessary to do a detailed inspection of the bulkheads and stiffeners below the seat rail attachments at FS 143.00 and FS 158.00.

**B. Special Tools**

- (1) None

**C. Access**

- (1) Remove the pilot and the copilot's seat. Refer to Chapter 25, [Flight Compartment - Maintenance Practices](#).
- (2) Remove floorboard access panels [231CL](#), [231DL](#), [232BC](#), [251PL](#), [251AL](#), [252PR](#), [232CR](#), and [232DR](#). Refer to Chapter 6, Access Plates and Panels - Description and Operation.

**D. Examine the Bulkheads and Stiffeners Below the Seat Rail Attachments.**

- (1) Examine the bulkheads and stiffeners for cracks at FS 143.00 and FS 158.00.
  - (a) Do a visual inspection of the bulkheads and stiffeners for cracks.
  - (b) Do a visual inspection of the bulkhead structure near the lightening holes for cracks.
  - (c) Do a visual inspection of the bulkhead structure near the intersections with the longitudinal bulkheads for cracks.
- (2) Examine the longitudinal bulkheads and stiffeners for cracks at FS 143.00 and FS 158.00.
  - (a) Do a visual inspection of the longitudinal bulkheads and stiffeners for cracks.
  - (b) Do a visual inspection of the bulkhead structure near the lightening holes for cracks.
  - (c) Do a visual inspection of the bulkhead structure near the intersections with the longitudinal bulkheads for cracks.
- (3) If there are cracks in the bulkheads, replace the bulkheads. Refer to the Model 208 Structural Repair Manual.
- (4) If there are no cracks, restore access.

**E. Restore Access**

- (1) Install floorboard access panels [231CL](#), [231DL](#), [232BC](#), [251PL](#), [251AL](#), [252PR](#), [232CR](#), and [232DR](#). Refer to Chapter 6, Access Plates and Panels - Description and Operation.
- (2) Install the pilot and the copilot's seat. Refer to Chapter 25, [Flight Compartment - Maintenance Practices](#).

**END OF TASK**

## SEAT RAILS - INSPECTION/CHECK

### 1. General

A. This section has the inspections and checks necessary to keep the seat rails in a serviceable condition.

#### TASK 53-25-00-220

### 2. Seat Rails and Attachment Structure Detailed Inspection

#### A. General

- (1) This task includes the Supplemental Inspection Document (SID) requirements necessary to do a detailed inspection of the seat rails.

#### B. Special Tools

- (1) None

#### C. Access

- (1) Remove the pilot and the copilot's seat. Refer to Chapter 25, [Flight Compartment - Maintenance Practices](#).
- (2) Remove floorboard access panels [231AL](#), [231BL](#), [231CL](#), [231DL](#), [232BC](#), [212FR](#), [216BC](#), [232AR](#), [232BR](#), [232AC](#), [232CR](#), and [232DR](#). Refer to Chapter 6, [Access Plates and Panels - Description and Operation](#).
- (3) Remove the passenger seats and floor covering. Refer to Chapter 25, [Passenger Seats - Maintenance Practices](#).
- (4) Remove necessary floor panels adjacent to passenger seat rails to examine structure beneath the passenger seat rails for cracks. Refer to Chapter 6, [Access Plates and Panels - Description and Operation](#).

#### D. Examine the Pilot and Copilot's Seat Rails and Attachment Structure.

- (1) Do a visual inspection of the pilot and copilot's seat rails for cracks or holes that are stretched more than 0.21 inch (5.33 mm).
  - (a) If any of the seat rails have cracks or holes that are stretched more than 0.21 inch (5.33 mm), replace the seat rails. Refer to Service Kit SK208-138 and [Seat Rails - Maintenance Practices](#).
  - (b) If there are no cracks or holes that are stretched more than 0.21 inch (5.33 mm), continue with the inspection. Refer to [Figure 601](#)
- (2) Do a visual inspection of the structure beneath the pilot and copilot's seat rails for cracks.
  - (a) If there are no cracks in the structure beneath the pilot or copilot's seat rails, restore access.
  - (b) If the structure beneath the pilot and copilot's seats has cracks, repair the structure. Refer to Service Kit SK208-138 and the Model 208 Structural Repair Manual.

#### E. Examine the Passenger Seat Rails and Attachment Structure.

- (1) Do a visual inspection of the passenger seat rails for cracks.
  - (a) If any of the seat rails have cracks, replace the seat rails. Refer to [Seat Rails - Maintenance Practices](#).
  - (b) If there are no cracks in the passenger seat rails, continue with the inspection.
- (2) Do a visual inspection of the structure under the passenger seat rails for cracks.
  - (a) If the structure under the passenger seats has cracks, repair the structure. Refer to the Model 208 Structural Repair Manual.
  - (b) If there are no cracks in the structure under the passenger seat rail restore access.

#### F. Restore Access

- (1) Install access panels [231AL](#), [231BL](#), [231CL](#), [231DL](#), [232BC](#), [212FR](#), [216BC](#), [232AR](#), [232BR](#), [232AC](#), [232CR](#), and [232DR](#). Refer to Chapter 6, [Access Plates and Panels - Description and Operation](#).
- (2) Install the pilot and the copilot's seat. Refer to Chapter 25, [Flight Compartment - Maintenance Practices](#).
- (3) Install floor panels adjacent to passenger seat rails that were removed. Refer to Chapter 6, [Access Plates and Panels - Description and Operation](#).
- (4) Install the passenger seats and floor covering. Refer to Chapter 25, [Passenger Seats - Maintenance Practices](#).

#### END OF TASK

#### TASK 53-25-00-221

### 3. Bulkheads and Stiffeners Below the Seat Rail Attachments at FS 143.00 and FS 158.00 Detailed Inspection

#### A. General

- D. Do a Special Detailed Inspection of the Fuselage Engine Mount Fittings.
  - (1) Do a nondestructive testing (NDT) inspection of the four truss assembly attachment points. Refer to the Model 208 Nondestructive Testing Manual, Part 6, Eddy Current, [Fuselage Engine Mount Fittings](#).
  - (2) Do a visual inspection of the gusset for cracks around the engine truss assembly attachment points.
  - (3) Do a visual inspection of the flange rings for cracks.
  - (4) Do an NDT inspection of the upper engine mount attachment. Refer to the Model 208 Nondestructive Testing Manual, Part 6, Eddy Current, [Fuselage Engine Mount Fittings](#).
  - (5) If no cracks are found, restore access.
  - (6) If cracks are found, repair or replace the damaged part(s). Refer to Chapter 71, [Engine Mount - Maintenance Practices](#) or the Model 208 Structural Repair Manual.
- E. Restore Access
  - (1) Install the engine cowl. Refer to [Engine Cowling and Nose Cap - Maintenance Practices](#).

**END OF TASK**

**TASK 53-10-00-251**

**15. Cargo and Passenger Door Doublers Special Detailed Inspection**

- A. General
  - (1) This task includes the Supplemental Inspection Document (SID) requirements necessary to keep the cargo and passenger door doublers in a serviceable condition.
- B. Special Tools
  - (1) None
- C. Do a Special Detailed Inspection of the Cargo and Passenger Door Doublers.
  - (1) Do a nondestructive testing (NDT) inspection of the upper passenger frame doublers and corners for cracks. Refer to the Model 208 Nondestructive testing Manual, Part 6, Eddy Current, [Cargo and Passenger Door Doublers](#).
  - (2) Do an NDT inspection of the upper cargo frame doublers and corners for cracks. Refer to the Model 208 Nondestructive testing Manual, Part 6, Eddy Current, [Cargo and Passenger Door Doublers](#).
  - (3) If no cracks are found, restore access.
  - (4) If cracks are found, repair or replace the damaged part(s).
    - (a) For the cargo doors, refer to Chapter 52, [Cargo Doors - Maintenance Practices](#) or the Model 208 Structural Repair Manual.

**END OF TASK**

**TASK 53-10-00-252**

**16. Fuselage to Wing Attach Fitting Lugs Special Detailed Inspection**

- A. General
  - (1) This task includes the Supplemental Inspection Document (SID) requirements necessary to keep the fuselage to wing attach fitting lugs in a serviceable condition.
- B. Special Tools
  - (1) None
- C. Access
  - (1) Remove the wing. Refer to Chapter 57, [Wings - Removal/Installation](#).
- D. Do a Special Detailed Inspection of the Fuselage to Wing Attach Fitting Lugs.
  - (1) Do a nondestructive testing (NDT) inspection of the bolt holes in the attach fittings lugs. Refer to the Model 208 Nondestructive testing Manual, Part 6, Eddy Current, [Fuselage to Wing Attach Fitting Lugs](#).
  - (2) If no cracks are found, restore access.
  - (3) If cracks are found, replace the fittings.
- E. Restore Access
  - (1) Install the wing. Refer to Chapter 57, [Wings - Removal/Installation](#).

**END OF TASK**





# **EMERGENCY EQUIPMENT LIST INSPECTION & MONITOR**

**PT. SMART CAKRAWALA  
AVIATION  
DEPARTMENT TEKNIK  
Form: SCA/MTC/023**

<b>DATE</b> :	<b>A/C REG</b> : PK-SNI
<b>A/C TYPE</b> : 208B Grand Caravan <b>EX</b>	<b>CHECKER</b> : <b>SIGN:</b>

No.	Description	P/N	S/N	Next Insp.	Remarks
1	Pilot Life Vest				
2	Co-Pilot Life Vest				
3	Pax Life Vest				
4	Pax Life Vest				
5	Pax Life Vest				
6	Pax Life Vest				
7	Pax Life Vest				
8	Pax Life Vest				
9	Pax Life Vest				
10	Pax Life Vest				
11	Pax Life Vest				
12	Pax Life Vest				
13	Firt Aid Kit				
14	Crash Axe Installed				
15	Fire Extinguisher				
16	Life Raft (If Installed)				
17	Survival Kit (If Installed)				
<b>OTHERS</b>					



## Additional Work Sheet

### Inspection Doc.15,01,MINOR

Aircraft Registration: **PK-SNI**

WO# Nr: **WO/022-SNI/IX/2022**

## Parts Used Sheet

### Special Tool Used

[illegible]



## Additional Work Sheet

### Inspection Doc.15,01,MINOR

Aircraft Registration: **PK-SNI**

WO# Nr: WO/022 SNI/IX/2022

## Parts Used Sheet

Part Used

[illegible]