

AIRCRAFT CHECK WORK SUMMARY
(Form: SCA/MTC/051)

DATE OF ISSUED	JOWO #	TYPE OF MAINTENANCE	DATE OF ACCOMPLISHED		
19 July 2022	WO/024-SNJ/VII/2022	Inspection Doc. 06 & Add Task			
A/C Type	Mfg. Serial Number	A/C Registration			
C208B	C208B-5640	PK-SNJ			
AIRCRAFT DATA					
Subject	Pos #	Serial Number (SN)	TTSN/TCSN		
Engine	#1	PCE-VA0738			
	#2	-			
Propeller/Rotor	#1	210140			
	#2	-			
Landing Gear	NLG				
	LH MLG				
	RH MLG				
PACKAGE COVERED					
No	Subject	Qty	Remark		
1	Non-Routine Card	1	NRC-001		
2	Inspection Card	1			
3	Work Order	1			
4	Summary Inspection List	1			
5	Material and Tool List	-			
6	Escalation form	-			
7	CRS (SMI / Unscheduled Maintenance)	1			
INSPECTION CARD (IC) LIST (Finding during maintenance)					
No	Taskcard Ref	Subject	Status		Name/ Sign & Stamp
			Open	Close	
<u>IC-001</u>					
<u>IC-002</u>					
<u>IC-003</u>					
<u>IC-004</u>					
<u>IC-005</u>					
<u>IC-006</u>					

<u>IC-007</u>					
<u>IC-008</u>					
<u>IC-009</u>					
<u>IC-010</u>					
<u>IC-011</u>					
<u>IC-012</u>					
<u>IC-013</u>					
<u>IC-014</u>					
<u>IC-015</u>					

Prepared by :
Technical Support

Checked by :
Chief Maintenance

Verified by :
Chief Inspector

Approved by :
Technical Manager



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PT. SMART CAKRAWALA AVIATION

CERTIFICATE RETURN TO SERVICE

SCHEDULED MAINTENANCE INSPECTION (CRS-SMI)

A/C TYPE : CESSNA 208B

TTSN :

A/C REG : PK-SNJ

TCSN :

MSN : C208B-5640

DATE :

TYPE OF INSPECTION : INSPECTION DOC. 06 & ADD TASK

DUE AT : 1000 FH

REF : MP C208B REV. 12

EXCEPTION

AUTHORIZED PERSON

I hereby certify that this aircraft has been maintained accordance with CASR and Maintenance Program.
Aircraft safe and airworthy for flight

NAME	CAT	AMEL/OTR NO	SIGN&STAMP	DATE
	AIRFRAME & POWER PLANT			
	EIRA			

THE NEXT DUE TYPE OF INSPECTION :

DUE AT :


Form: SCA/MTC/049



SUMMARY INSPECTION ITEMS (Form: SCA/MTC/050)

WO Ref: WO/024-SNJ/VII/2022

NO.	TASK CARD NO.	DESCRIPTION	DATE	EST MHR	NAME	STAMP
1	B03	PT6A-140 ENGINE GROUND RUN PERFORMANCE				
2	CHAPTER 12	INSPECTION DOCUMENT 06				
3	CHAPTER 51	ENGINE PT6A-140 100 HOURS INSPECTION/ MINOR INSPECTION				
4	CHAPTER 52	ENGINE PT6A-140 200 HOURS INSPECTION				
5	CHAPTER 53	ENGINE PT6A-140 200 HOUR/6 MONTHS INSPECTION				
6	CHAPTER 56	ENGINE PT6A-1000 HOURS INSPECTION				
7	SCA/MTC/0 23	EMERGENCY EQUIPMENT CHECK				
8	NRC-001	STARTER GENERATOR REPLACEMENT				

	INSPECTION CARD (Form: SCA/MTC/ 048)	TECHNICAL DEPARTMENT
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1. CARD #	2. JO/WO #	3. ORIGINATOR	4. CARD REF	5. DATE
6. A/C REG/MSN	7. A/C TYPE	8. TRADE	12. VENDOR ORDER #	
9. ZONE	10. STA	11. MTC TYPE		

13. DESCRIPTION/DEFECT-IF FINDING OF CPCP INSPECTION, PLEASE COMPLETE SET. 20	14 PPC/ENG	15 DATE

16. CORRECTIVE ACTION	17 MECH	18 ENG. LIC	19 DATE
Performed at A/C TT : A/C TC /LDG :			

20. CORROSION INFORMATION					
LOCATION	CAUSE OF DAMAGE				
	<input type="checkbox"/> Environment				
	<input type="checkbox"/> Internal Leakage				
CORROSION <input type="checkbox"/> Isolated <input type="checkbox"/> Widespread	<input type="checkbox"/> Chemical Spill				
CORROSION LVL <input type="checkbox"/> 1 <input type="checkbox"/> 2 <input type="checkbox"/> 3	<input type="checkbox"/> LAV/Galley Spill				
PROPOSED ACTION <input type="checkbox"/> Doublers	<input type="checkbox"/> Blocked Drain				
<input type="checkbox"/> Others	<input type="checkbox"/> Wet Insulation Blanket				
	<input type="checkbox"/> Other				

21. If the defect is RII, Please Sign this card finally by RII Inspector	INSP	DATE
NOTICE OF INSPECTOR		

22. PARTS REQUIRED						
PART DESCRIPTION	PART NO	QTY	SERIAL NO		STATUS	
			ON	OFF	CLOSE	OPEN

23. TOOLS REQUIRED			
DESCRIPTION	PART NO. / MODEL	NEXT CALIBRATION DATE	STATUS



NON ROUTINE CARD
(Form: SCA/MTC/047)

1. JO/WO #	2. DATE	3. MTC TYPE	4. A/C REG/MSN
WO/042-SNJ/VII//2022		REPLACEMENT COMPONENT	PK-SNJ-5640
5. CARD #	6. ATA SPEC	7. TRADE	8. STA
#001	80		
9. ZONE	10. PANEL	-	
ENGINE			

11. DESCRIPTION			
PERFORM STARTER GENERATOR REPLACEMENT REPLACEMENT P/N: 300SGL145Q-1 S/N: S00480			
REFERENCE	<input checked="" type="checkbox"/> AMM	<input type="checkbox"/>	<input type="checkbox"/> OTHER
RII (*)	<input type="checkbox"/> Y	<input type="checkbox"/> N	MHR :

12. RESULT		MECH	ENG	INSP (*)
Performed at A/C TT : A/C TC /LDG :				
FINDING	<input type="checkbox"/> Y	<input type="checkbox"/> N	ACT MHR :	DATE/TIME (DD/MM/YY)
INSPECTION CARD (IC) #				

13. PARTS REQUIRED				
DESCRIPTION	PART NO	QTY	REMARK	
			STOCK	STATUS

14. TOOLS REQUIRED			
DESCRIPTION	PART NO / MODEL	NEXT CALIBRATION DATE	STATUS



MAINTENANCE PROGRAM CESSNA C208/C208B

Appendix A26 – Starter-Generator Replacement

Reg. Mark : PK - SNJ
MSN : 5640
TSN / CSN :

Date :
Station :
WO No. : WO/042-SNJ/VII//2022

NO.	ZONE	TASK	SIGNATURE	
			SIGN	STAMP
01	130	For engine PT6A-114A if there is an engine starting fault or an electrical generation defect, inspect the starter generator drive spline and the main oil filter. Refer to EMM Chapter 72-00-00 12Y		
02	130	For engine PT6A-140 if there is an engine starting fault or an electrical generation defect, inspect the starter generator drive spline and the main oil filter. Refer to EMM Task 05-50-00-210-824		
*** End of Starter Generator Replacement Inspection Items ***				

PERSONNEL PARTICIPATING IN THIS INSPECTION			
NAME	POSITION	SIGNATURE	LICENSE NUMBER

RETURN TO SERVICE

The work recorded above has been carried out in accordance with the requirements of the Civil Aviation Safety Regulation for the time being in force and in that respect the aircraft is considered fit for Release to Service.

Name : _____ Stamp : _____
Signature : _____ Place/Date : _____

STARTER/GENERATOR - REMOVAL/INSTALLATION**1. General**

- A. This section gives removal and installation information for all starter/generators used on the airplane.

2. Starter/Generator Removal and Installation

- A. Remove the Starter/Generator (Refer to [Figure 401](#)).

NOTE: Two mechanics are required to properly remove or install the starter/generator.

CAUTION: Make sure the starter/generator drive shaft is aligned with and concentric to the armature shaft. Slight misalignment and/or binding of the starter/generator drive can reduce the unit's service life.

- (1) Remove the left and right upper cowl doors. Refer to Chapter 71, [Engine Cowling and Nose Cap - Maintenance Practices](#).
- (2) Remove all external power from the airplane, make sure the battery switch is in the OFF position, and disconnect the battery from the airplane electrical system.
- (3) Remove the cover from the terminal block.
- (4) Put an identification tag on each of the electrical leads for later identification and remove the terminal nuts.
- (5) Remove the speed sensor circuit connector.

NOTE: Removal of the A/C drive unit is necessary for access to the starter generator on airplanes before 208000505 and 208B002025 that have a 200 AMP starter generator option installed.

- (6) Loosen the clamp that holds the cooling air blast hose on the starter/generator and remove the hose.

NOTE: Two mechanics are required to properly remove or install the starter/generator. One mechanic is to hold the starter/generator in position to keep the mounting surfaces flush with the quick attach/detach (QAD) adapter pad. This keeps the starter/generator aligned while the other mechanic loosens and removes the V-band clamp.

CAUTION: Hold the starter/generator in place to prevent damage to the splined drive shaft before you do the following step.

- (7) Loosen the V-band that holds the starter/generator to the quick attach/detach QAD adapter.
- (8) Carefully remove the starter/generator from the QAD adapter pad so that the starter/generator drive spline is not put into a bind.
- (9) Remove the QAD adapter as necessary.
 - (a) Remove the nuts that hold the QAD adapter to the engine accessory gearbox and remove the adapter.
 - (b) Discard the gasket.
- (10) Use a cloth that is damp with MIL-PRF-680 or an equivalent solvent to clean the starter-generator splines.
- (11) Use a 10X magnifying glass to examine for signs of electrical arcing damage (in the form of pitting).

NOTE: If there are signs of arcing on the starter-generator drive splines, refer to Cessna SNL07-16 and P&WC S.I.L NO. Gen-PT6-024 for additional information and inspection requirements.

NOTE: If the Starter-Generator was removed for an electrical fault, refer to the Pratt and Whitney Canada PT6A Maintenance Manual 05.50.00 unscheduled inspection section- Starter-Generator Replacement.

- B. Install the Starter/Generator (Refer to [Figure 401](#)).

- (1) Do the following steps before you install the starter/generator. Make sure:
 - (a) There are no burrs or foreign objects on the starter/generator shaft.
 - (b) The starter/generator guide pins are clean and not bent or damaged.
 - (c) The mounting surfaces of the starter/generator and the QAD adapter pad are clean and do not have any burrs.
 - (d) The QAD adapter is fastened to the engine transfer case correctly.
 - (e) The QAD adapter pad guide pin holes does not have any burrs or foreign objects, and that they are in good condition.

NOTE: For a 300-Amp Starter/Generator installation, the QAD adapter must be located with the internal machined circular recesses on the top.

- (2) Install the QAD adapter onto engine accessory gearbox with a new gasket, and install nuts as necessary.
- (3) Install a new O-ring around the groove on the splined drive shaft.
- (4) With the T-bolt unlatched, put the V-band on the starter/generator between the mounting flange and the terminal block.

NOTE: Two mechanics are required to properly remove or install the starter/generator. One mechanic is to hold the starter/generator in position to keep the mounting surfaces flush with the quick attach/detach QAD adapter pad. This keeps the starter/generator aligned while the other mechanic installs and tightens the V-band clamp.

CAUTION: The spline drive shaft must stay aligned with and concentric to the armature. If the starter/generator is allowed to be installed with the drive shaft out of position, excessive vibration and damage may develop during operation.

- (5) Carefully look at the spline drive shaft and the armature shaft interface plates. If the drive shaft looks to be out of position, lightly tap on the spline drive shaft with a plastic mallet to move it to a full concentric position. [Figure 401](#).
- (6) Carefully engage the spline drive shaft with the engine spline.

CAUTION: Keep the starter/generator flush up against the adapter during installation. Do not let the unit hang loosely without the V-band being latched because too much load on the drive shaft shear section may cause damage.

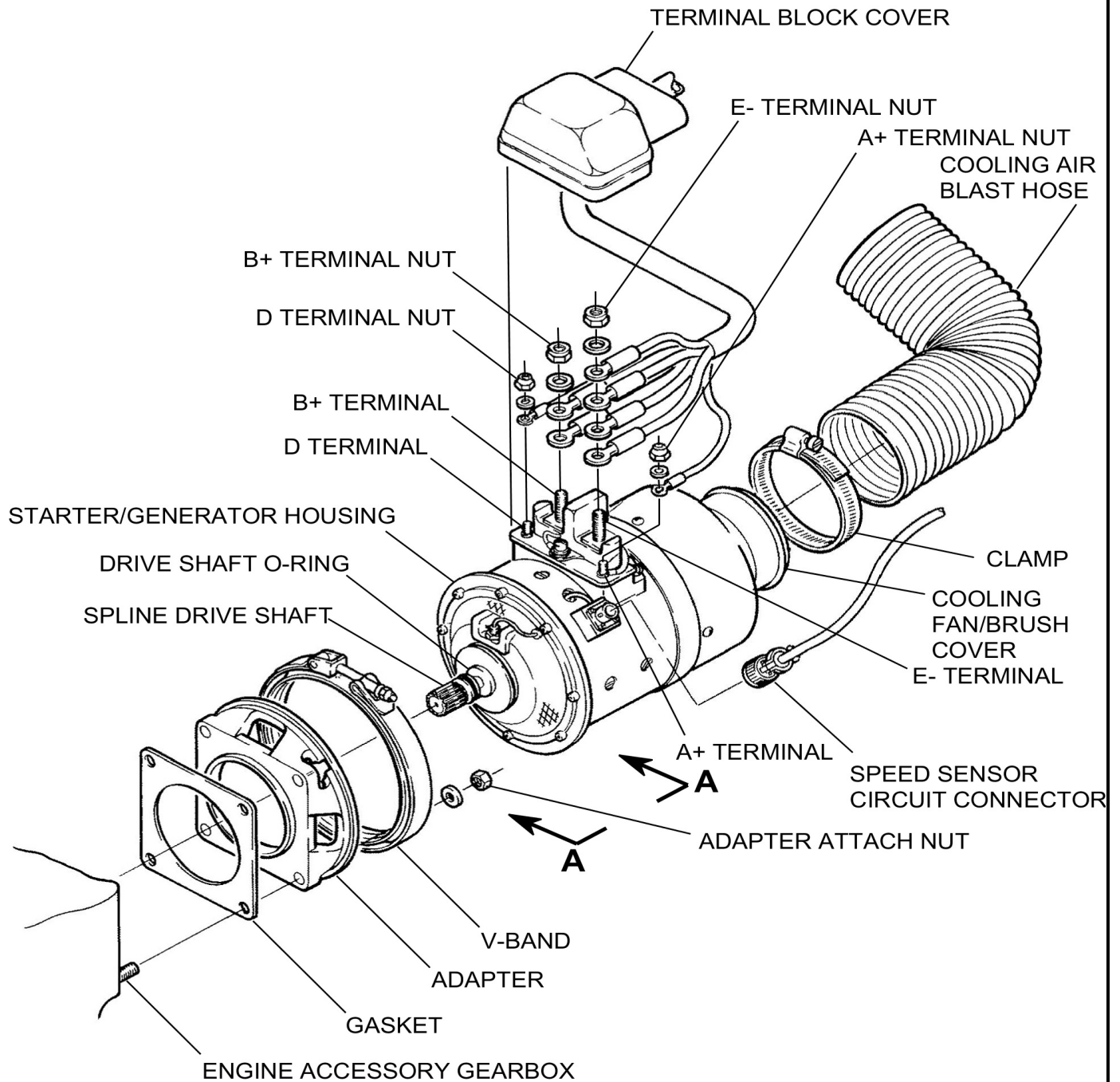
- (7) Make sure the dowel pins are engaged.
- (8) Put the V-band over the mating flanges and latch.
- (9) Tap the V-band at several places with a rubber mallet to make sure that there is correct alignment of the spline drive shaft and the armature shaft, and tighten the T-bolt nut to two-thirds the recommended torque.

NOTE: The correct torque value is stamped on the V-band.

- (10) Tap the V-band repeatedly with the rubber mallet and tighten the T-bolt nut to the recommended torque.
- (11) Install the cooling air blast hose with the clamp on the starter/generator.
- (12) Tighten the cooling air blast hose clamp.
- (13) Connect the speed sensor cable connector to the starter/generator.
- (14) Install the electrical cables in the same relationship to the terminal posts as you tagged them during the removal procedure, and install the nuts.
- (15) Put the cover in place over the terminal block.
- (16) Reconnect the battery to the airplane electrical system.
- (17) Install the left and right upper cowling doors. Refer to Chapter 71, [Engine Cowling and Nose Cap - Maintenance Practices](#).

Figure 401 : Sheet 1 : Starter/Generator Installation

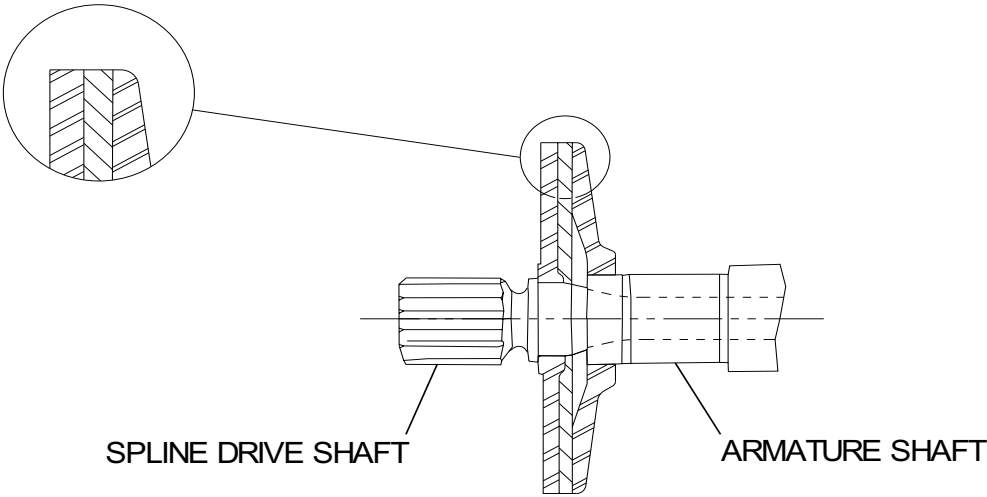
A21663



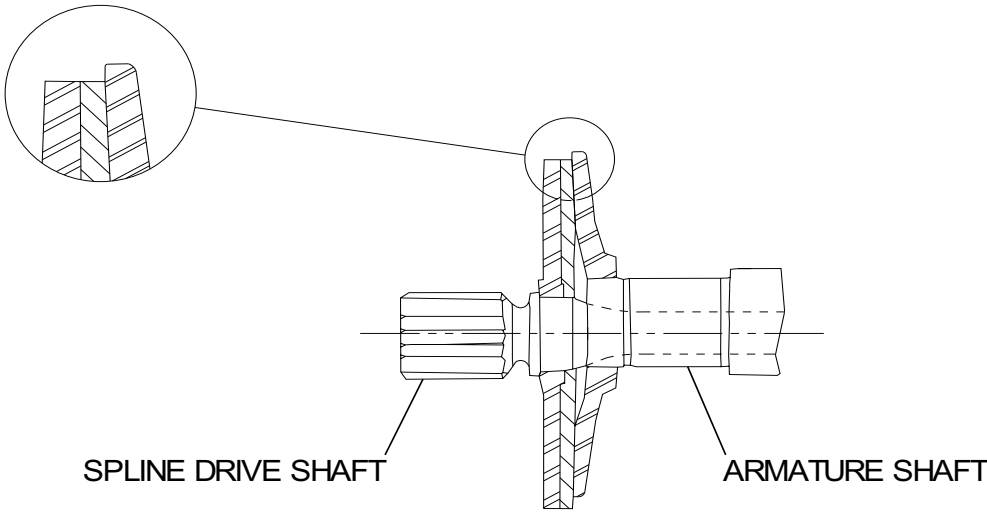
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Figure 401 : Sheet 2 : Starter/Generator Installation

A68000



VIEW A-A
DRIVE SHAFT CONCENTRIC WITH ARMATURE SHAFT



VIEW A-A
DRIVE SHAFT OUT OF POSITION



MAINTENANCE PROGRAM CESSNA C208/C208B

Appendix B03 – PT6A-140 Engine Run Performance Sheet

Reg. Mark : PK - SNJ

WO/FML No. : WO/042-SNJ/VII//2022

PRE – INSPECTION	
Location	
Date	
Cycle	
Filed Barometric	
OAT	
Altitude	

POST – INSPECTION	
Location	
Date	
Cycle	
Filed Barometric	
OAT	
Altitude	

PRE – INSPECTION		
	Target	Actual
Tq		
Np		
ITT	°C	°C
Ng	%	%
Wf		
Oil Press		°C
Oil Temp		°C
Start Temp		°C

POST – INSPECTION		
	Target	Actual
Tq		
Np		
ITT	°C	°C
Ng	%	%
Wf		
Oil Press		°C
Oil Temp		°C
Start Temp		°C

Engine Run Up Checks							
Inertial <input type="checkbox"/>	EPL <input type="checkbox"/>	OVG <input type="checkbox"/>	Stby Alt <input type="checkbox"/>	BOV <input type="checkbox"/>	Brake <input type="checkbox"/>	Randomn <input type="checkbox"/>	
NOTE: 1. Brake system at Torque 2000 ft-lbs. 3. EPL check can't exceed 4% Ng per second. 5. Low idle at 55.5 - 57% 40Amps. 2. Inertial Separator at Torque 400 ft-lbs. 4. Standby Alt at 80% Ng. 6. High idle at 64 - 66% Ng 40Amps							

Engine Performance Target Table (Cessna C208B EX)

OAT (°C)	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41
Tq (ft.lbs)	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397
Np	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900
ITT (°C)	835	837	839	841	841	841	841	841	841	842	843	844	846	846	846
Ng (%)	102.7	102.7	102.7	102.7	102.7	102.7	102.7	102.6	102.6	102.6	102.6	102.6	102.6	102.6	102.5
WF (PPH)	578	578	578	578	578	578	578	570	565	565	560	560	555	548	548

Note:

1. Make sure that inertial separator in normal condition, no bleed air extracted from the engine and air condition OFF.
2. This table only applies to altitude 0-500 feet MSL. For higher altitude, refer to EMM 72-00-00.
3. Max fuel flow is 580 lb/hr fuel flow is not more than 15 lbs/hr higher than the value shown in table.
4. If parameters are outside the target performance table to EMM chapter 71-00-00.

REMARKS:

PERFORMED BY

Name	Sign & Stamp	Date	Location



MAINTENANCE PROGRAM CESSNA C208/C208B

Chapter 12 – Inspection Document 06

Reg. Mark	: PK - SNJ	Date	:	
MSN	: 5640	Station	:	
TSN / CSN	:	WO No.	:	WO/042-SNJ/VII//2022

ITEM CODE NO.	ZONE	TASK	SIGNATURE	
			SIGN	STAMP
A110001	ALL	Interior and Exterior Placard and Decal Detailed Inspection Task 11-00-00-220		
D121001	121	Brake System Servicing Task 12-10-01-610		
D121003	710	Shimmy Damper Servicing Task 12-10-01-611		
C122101	700	Landing Gear Lubrication Task 12-21-03-640		
B236001	343 375 376 571 671	Static Discharge System Functional Check Task 23-60-00-720		
A255101	251 252 255 256 257 258	Cargo Nets Detailed Inspection Task 25-51-00-220		
C270001	215 216 226 373 374 503 525 603 625	Flight Controls Lubrication Task 27-00-00-640		
B273101	211 212 503	Stall Warning System Operational Check Task 27-31-00-710		
C275001	525 527 625 627	Flap Tracks and Rollers Lubrication Task 27-50-00-640		
B281001	575 675	Fuel Vent Line Float Valve Operational Check Task 28-10-03-710		
A281001	521 621	Fuel Filler Assembly Detailed Inspection Task 28-10-01-220		
B301003	122 AUX	Bleed Air Pressure Regulator Functional Check (without TKS and not incorporating CAB93-2) Task 30-10-00-720		
B322001	710	Shimmy Damper Functional Check Task 32-20-02-720		



MAINTENANCE PROGRAM **CESSNA C208/C208B**

Chapter 12 – Inspection Document 06

ITEM CODE NO.	ZONE	TASK	SIGNATURE	
			SIGN	STAMP
B341101	AUX	Pitot Tube Heaters Operational Check Task 34-11-00-710		
A353001	256	Oxygen Mask Detailed Inspection Task 35-30-00-220		
B761003	AUX	Emergency Power Lever Annunciator Light (EPL) Operational Check Task 76-10-01-710		
*** End of Inspection Document 06 Items ***				

PERSONNEL PARTICIPATING IN THIS INSPECTION			
NAME	POSITION	SIGNATURE	LICENSE NUMBER

RETURN TO SERVICE

The work recorded above has been carried out in accordance with the requirements of the Civil Aviation Safety Regulation for the time being in force and in that respect the aircraft is consider fit for Release to Service.

Name : _____ Stamp : _____
Signature : _____ Place/Date : _____



MAINTENANCE PROGRAM CESSNA C208/C208B

Chapter 51 – Engine PT6A-140 100 Hours/Minor Inspection

Reg. Mark	:	PK - SNJ	Date	:	
MSN	:	5640	Station	:	
TSN / CSN	:		WO No.	:	WO/042-SNJ/VII//2022

ITEM CODE NO.	ZONE	TASK	SIGNATURE		DATE
			SIGN	STAMP	
F710001	130	Do a check of the FCU manual override system for static operation. For the engines installed with a manual override system only.			
F710003	130	Do a compressor performance recovery wash			
F720000	130	Do a visual inspection of the Control Linkages and wiring.			
F720001	130	Do a visual inspection of the engine exhaust duct welds.			
F720002	130	Do a visual inspection of the engine exhaust duct for cracks.			
F720003	130	Do a visual inspection of the gas generator case and the center fireseal.			
F720004	130	Do a visual inspection of the rear fireseal mounting ring.			
F722001	130	Do a visual inspection of the air inlet screen.			
F723000	130	Do a visual inspection with a mirror or a borescope inspection of the First-stage Compressor Rotor and the inlet case for corrosion			
F725005	130	Do a detailed inspection of the turbine exhaust duct.			
F731002	130	Do a check for the fuel pump installation and leaks.			
F731003	130	Do a check of the oil-to- fuel heater installation			
F731035	130	Do a visual Inspection of the Fuel - Oil Heat Exchanger Fuel Filter Element (CLEANING / REPLACEMENT) P/N OFF: _____ P/N ON: _____			

Chapter 51 – Engine PT6A-140 100 Hours/Minor Inspection

ITEM CODE NO.	ZONE	TASK	SIGNATURE		DATE
			SIGN	STAMP	
F731006	130	Do a check of the drain valve for installation and leaks			
		NOTE: There is no need to remove the drain valve if there is no leaks.			
F731007	130	Do a check of the flow divider for installation and leaks.			
F731008	130	Do a visual inspection of the P3 filter and drain valve.			
F731015	130	Do a visual inspection of Fuel Pump outlet filter. (CLEANING / REPLACEMENT) P/N OFF: _____ P/N ON: _____			
F731018	130	Clean or replace the P3 filter based on condition, service experience or environment. Note: If corrossions are found, replace filter.			
F732001	130	Do a check of the FCU for installation, linkages and pneumatic tubes.			
F792000	130	Inspect and clean oil filter for debris.			
*** End of Engine PT6A-140 100 Hours Inspection Items ***					

PERSONNEL PARTICIPATING IN THIS INSPECTION			
NAME	POSITION	SIGNATURE	LICENSE NUMBER

RETURN TO SERVICE			
<p>The work recorded above has been carried out in accordance with the requirements of the Civil Aviation Safety Regulation for the time being in force and in that respect the aircraft is consider fit for Release to Service.</p>			
Name	: _____	Stamp	: _____
Signature	: _____	Place/Date	: _____



MAINTENANCE PROGRAM CESSNA C208/C208B

Chapter 52 – Engine PT6A-140 200 Hours Inspection

Reg. Mark	:	PK - SNJ	Date	:	
MSN	:	5640	Station	:	
TSN / CSN	:		WO No.	:	WO/042-SNJ/VII//2022

ITEM CODE NO.	ZONE	TASK	SIGNATURE	
			SIGN	STAMP
F793001	130	Do a visual inspection of the chip detector for debris.		
*** End of Engine PT6A-140 200 Hours Inspection Items ***				

PERSONNEL PARTICIPATING IN THIS INSPECTION			
NAME	POSITION	SIGNATURE	LICENSE NUMBER

RETURN TO SERVICE

The work recorded above has been carried out in accordance with the requirements of the Civil Aviation Safety Regulation for the time being in force and in that respect the aircraft is consider fit for Release to Service.

Name	:		Stamp	:	
Signature	:		Place/Date	:	



MAINTENANCE PROGRAM CESSNA C208/C208B

Chapter 53 – Engine PT6A-140 200 Hours or 6 Months Inspection

Reg. Mark	:	PK - SNJ	Date	:	
MSN	:	5640	Station	:	
TSN / CSN	:		WO No.	:	WO/042-SNJ/VII//2022

ITEM CODE NO.	ZONE	TASK	SIGNATURE	
			SIGN	STAMP
F793000	130	Do a visual inspection of the AGB internal scavenge oil pump inlet screen		
*** End of Engine PT6A- 140 200 Hours or 6 Months Inspection Items ***				

PERSONNEL PARTICIPATING IN THIS INSPECTION			
NAME	POSITION	SIGNATURE	LICENSE NUMBER

RETURN TO SERVICE

The work recorded above has been carried out in accordance with the requirements of the Civil Aviation Safety Regulation for the time being in force and in that respect the aircraft is consider fit for Release to Service.

Name	:		Stamp	:	
Signature	:		Place/Date	:	



MAINTENANCE PROGRAM CESSNA C208/C208B

Chapter 56 – Engine PT6A-140 1000 Hours Inspection

Reg. Mark	:	PK - SNJ	Date	:	
MSN	:	5640	Station	:	
TSN / CSN	:		WO No.	:	WO/042-SNJ/VII//2022

ITEM CODE NO.	ZONE	TASK	SIGNATURE	
			SIGN	STAMP
F731028	130	Replace the P3 filter. P/N OFF: _____ P/N ON: _____		
F792001	130	Remove and discard the oil filter element. Install a new oil filter element. P/N OFF: _____ P/N ON: _____		
*** End of Engine PT6A-140 1000 Hours Items ***				

PERSONNEL PARTICIPATING IN THIS INSPECTION			
NAME	POSITION	SIGNATURE	LICENSE NUMBER

RETURN TO SERVICE

The work recorded above has been carried out in accordance with the requirements of the Civil Aviation Safety Regulation for the time being in force and in that respect the aircraft is consider fit for Release to Service.

Name	:	_____	Stamp	:	_____
Signature	:	_____	Place/Date	:	_____



EMERGENCY EQUIPMENT LIST INSPECTION & MONITOR

**PT. SMART CAKRAWALA
AVIATION**
DEPARTMENT TEKNIK
Form: SCA/MTC/023

DATE :	A/C REG : PK-SNJ
A/C TYPE : C208B	CHECKER : SIGN:

No.	Description	P/N	S/N	Next Insp.	Remarks
1	Pilot Life Vest				
2	Co-Pilot Life Vest				
3	Pax Life Vest				
4	Pax Life Vest				
5	Pax Life Vest				
6	Pax Life Vest				
7	Pax Life Vest				
8	Pax Life Vest				
9	Pax Life Vest				
10	Pax Life Vest				
11	Pax Life Vest				
12	Pax Life Vest				
13	Firt Aid Kit				
14	Crash Axe Installed				
15	Fire Extinguisher				
16	Life Raft (If Installed)				
17	Survival Kit (If Installed)				
OTHERS					



Additional Work Sheet

Inspection Doc. 06 & Add Task

Aircraft Registration: **PK-SNJ**

WO# Nr: **WO/024-SNJ/VII/2022**

Parts Used Sheet

Special Tool Used

[illegible]



Additional Work Sheet

Inspection Doc. 06 & Add Task

Aircraft Registration: **PK-SNJ**

WO# Nr: WO/024-SNJ/VII/2022

Parts Used Sheet

Part Used

[illegible]