

AIRCRAFT CHECK WORK SUMMARY
(Form: SCA/MTC/051)

DATE OF ISSUED	JO/WO #	TYPE OF MAINTENANCE	DATE OF ACCOMPLISHED		
10 Apr 2023	WO/002-SNV/IV/2023	Propeller Assy Replacement			
A/C Type	Mfg. Serial Number	A/C Registration			
C208B	C208B-5551	PK-SNV			
AIRCRAFT DATA					
Subject	Pos #	Serial Number (SN)	TTSN/TCSN		
Engine	#1	PCE-VA0607			
	#2	-			
Propeller/Rotor	#1	181160			
	#2	-			
Landing Gear	NLG				
	LH MLG				
	RH MLG				
PACKAGE COVERED					
No	Subject	Qty	Remark		
1	Non-Routine Card	1			
2	Inspection Card				
3	Work Order	1			
4	Summary Inspection List	1			
5	Material and Tool List	-			
6	Escalation form	-			
7	CRS (SMI / Unscheduled Maintenance)	1			
INSPECTION CARD (IC) LIST (Finding during maintenance)					
No	Taskcard Ref	Subject	Status		Name/ Sign & Stamp
			Open	Close	
<u>IC-001</u>					
<u>IC-002</u>					
<u>IC-003</u>					
<u>IC-004</u>					
<u>IC-005</u>					
<u>IC-006</u>					

<u>IC-007</u>					
<u>IC-008</u>					
<u>IC-009</u>					
<u>IC-010</u>					
<u>IC-011</u>					
<u>IC-012</u>					
<u>IC-013</u>					
<u>IC-014</u>					
<u>IC-015</u>					

Prepared by :
Technical Support

Checked by :
Chief Maintenance

Verified by :
Chief Inspector

Approved by :
Technical Manager



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SUMMARY INSPECTION ITEMS
(Form: SCA/MTC/050)

WO Ref: WO/040-SNV/IV/2023

NO.	TASK CARD NO.	DESCRIPTION	DATE	EST MHR	NAME	STAMP
1	NRC-001	REPLACEMENT OF PROPELLER ASSY 4HFR34C778 REF EO NO. 002/EO/TEK-TS/IV/2023				



PT. SMART CAKRAWALA AVIATION

CERTIFICATE RETURN TO SERVICE

SCHEDULED MAINTENANCE INSPECTION (CRS-SMI)

A/C TYPE : CESSNA 208B
A/C REG : PK-SNV
MSN : C208B-5551

TTSN :
TCSN :
DATE :

TYPE OF INSPECTION : PROPELLER ASSY REPLACEMENT :
DUE AT
REF : EO NO. 002/EO/TEK-TS/IV/2023

EXCEPTION


AUTHORIZED PERSON

I hereby certify that this aircraft has been maintained accordance with CASR and Maintenance Program.
Aircraft safe and airworthy for flight

NAME	CAT	AMEL/OTR NO	SIGN&STAMP	DATE
	AIRFRAME & POWER PLANT			
	EIRA			

THE NEXT DUE TYPE OF INSPECTION :
DUE AT :

Form: SCA/MTC/049

	INSPECTION CARD (Form: SCA/MTC/ 048)	TECHNICAL DEPARTMENT
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1. CARD #	2. JO/WO #	3. ORIGINATOR	4. CARD REF	5. DATE
6. A/C REG/MSN	7. A/C TYPE	8. TRADE	12. VENDOR ORDER #	
9. ZONE	10. STA	11. MTC TYPE		


13. DESCRIPTION/DEFECT-IF FINDING OF CPCP INSPECTION, PLEASE COMPLETE SET. 20	14 PPC/ENG	15 DATE

16. CORRECTIVE ACTION	17 MECH	18 ENG. LIC	19 DATE
Performed at A/C TT : A/C TC /LDG :			

20. CORROSION INFORMATION					
LOCATION	CAUSE OF DAMAGE				
	<input type="checkbox"/> Environment				
	<input type="checkbox"/> Internal Leakage				
CORROSION <input type="checkbox"/> Isolated <input type="checkbox"/> Widespread	<input type="checkbox"/> Chemical Spill				
CORROSION LVL <input type="checkbox"/> 1 <input type="checkbox"/> 2 <input type="checkbox"/> 3	<input type="checkbox"/> LAV/Galley Spill				
PROPOSED ACTION <input type="checkbox"/> Doublers	<input type="checkbox"/> Blocked Drain				
<input type="checkbox"/> Others	<input type="checkbox"/> Wet Insulation Blanket				
.....	<input type="checkbox"/> Other				
21. If the defect is RII, Please Sign this card finally by RII Inspector				INSP	DATE
NOTICE OF INSPECTOR					

22. PARTS REQUIRED						
PART DESCRIPTION	PART NO	QTY	SERIAL NO		STATUS	
			ON	OFF	CLOSE	OPEN

23. TOOLS REQUIRED			
DESCRIPTION	PART NO. / MODEL	NEXT CALIBRATION DATE	STATUS

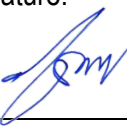
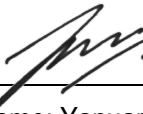
	TECHNICAL SUPPORT TECHNICAL DEPARTMENT ENGINEERING ORDER	002/TEK-TS/IV/2023	
		Rev. No	Original
		Rev. Date	10 Apr 2023


ENGINEERING ORDER

002/TEK-TS/IV/2023

REMOVAL & INSTALLATION OF PROPELLER MCCAULEY MODEL 4HFR34C778 Series ON CESSNA 208B GRAND CARAVAN

PT. SMART CAKRAWALA AVIATION

Prepared	Checked	Approved
Technical Support	Technical Manager	Chief Inspector
Signature: 	Signature: 	Signature: 
Name: Dwi M	Name: Istiono	Name: Yanuar A. F.
Date: 10 April 2023	Date: 10 April 2023	Date: 10 April 2023

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		Rev. No	Original
		Rev. Date	10 Apr 2023

SMART AVIATION ENGINEERING ORDER			
Aircraft Reg.: PK-SNV (208B5551)	Make/Model: C208B	No. El: 002/TEK-TS/IV/2023	Rev. No. : Original
Total Flight Hours:	Total Flight Cycle:	Date Issued : 10 April 2023	
Task Description : REMOVAL & INSTALLATION OF PROPELLER MCCAULEY MODEL 4HFR34C778 Series ON CESSNA 208B GRAND CARAVAN		Technical Data Reference : <u>MCCAULEY PROPELLER SYSTEMS Propeller Owner/Operator Information Manual C700/C750/C1000 Propeller Removal/Installation</u>	
Effectivity : CESSNA 208B EQUIPPED WITH PROPELLER MCCAULEY MODEL 4HFR34C778 Series			



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**SMART AVIATION
ENGINEERING ORDER**

1. Description.

This EO is issued, to perform removal & installation checklist Propeller Assembly maintenance practices the 4HFR34C778 Series Propeller on Cessna 208B Grand Caravan.

2. Aircraft Effectivity.

REGISTRATION	SERIAL NUMBER
PK-SNV	208B5551

3. Distribution :

TECHNICAL MANAGER	[√]	MATERIAL SUPPORT	[√]
SAFETY & QUALITY MANAGER	[√]	TECHNICAL SUPPORT	[√]
CHIEF INSPECTOR	[√]	FILE	[√]

4. Man Hours

18.0 man-hour to do the inspection

5. Material

A1633-72 Packing
A1639-32 Nut, Propeller

6. Special Tool Required

Tracking, Propeller
Adapter, Torque Wrench
Start Lock Release

7. Compliance

The Propeller model 4HFR34C778 Series have 4 of Blades, do a removal the propeller installed on Engine refer to accomplishment instruction task card, and install the Serviceable/New Propeller on the aircraft refer to accomplishment instruction task card.

8. Publications Affected.

None.



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**SMART AVIATION
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9. Accomplishment Instructions.

PROPELLER REMOVAL

Date : WO Number :
Part No. Propeller : 4HFR34C778- A/C Total Hours :
Serial No. Propeller : A/C Total :
Landing
Propeller Time : TSN: TSO:
Removed from A/C Reg. :

Description	Eng.	Rll	Remarks
A. REMOVE PROPELLER (Refer to Figure 01 to 04)			
1. Make sure the airplane propeller control lever has the FEATHER position selected and the propeller blades are in feather.			
NOTE: If the propeller control lever is not in the FEATHER position or the propeller blades are not in feather, consult the Pilot's Operating Handbook for instructions to put the propeller control in the feather position and to feather the propeller.			
2. Place a drip pan under the propeller mounting surface to catch any spilled engine oil when the propeller is removed.			
3. Remove the feedback bearing assembly from the feedback collar groove. Refer to the aircraft or engine manual for procedure.			
4. Remove the cowlings as required for access to the mounting nuts.			
5. Remove the cowlings as required for access to the mounting nuts.			
NOTE: Make sure the rods are inserted through the correct holes of the tool. The beta rods must be pulled directly forward.			
CAUTION: Do not draw the feedback collar too far forward as the beta rods will be damaged			
6. Turn the large threaded rod of the retractor tool clockwise to draw the feedback collar forward to allow access to the propeller mounting nuts.			
7. Break the torque on the propeller mounting nuts. (McCauley Torque Wrench Adapter part number B-5588 may be used to gain easier access to mounting nuts.)			
8. Install propeller sling and attach to hoist. If hoist is unavailable, propeller may be supported and lifted by hand. This will require			



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additional personnel.

CAUTION: The C700 propellers weigh approximately 120 pounds (54 kg), the C750 propellers approximately 140 pounds (64 kg). CI000 series propellers weigh 180 pounds (83 kg) or more

- A. Hoist straps must be a minimum of 4 inches {100 mm} wide
- B. The sling and hoist should have a weight limit rating at least twice the weight of the propeller that is to be removed.
- C. The straps of the propeller sling should be placed on two of the propeller blades at least 6 inches (152 mm) outboard of the propeller hub. Make sure you protect the deice boots or anti-ice shoes from potential propeller sling abrasion damage, if installed.
9. Take up the slack on the propeller sling and hoist, or have additional personnel support the propeller, and remove the nuts.
10. Carefully slide the propeller forward and remove from the engine
11. Place the propeller on a suitable support or propeller stand
 - a. Do not let the propeller be supported by the tips of the propeller blades or "stand" the propeller on the propeller blade tips
 - b. The stand should be able to hold the propeller securely and have a weight limit rating at least twice that of the removed propeller.
 - c. Make sure the propeller is protected from damaged while in the stand. All areas of the stand that come in contact with the propeller should be padded sufficiently to prevent abrasion damage to the propeller.
12. Install a protective cover over the propeller hub mounting flange.
13. Install a protective cover over the open end of the engine propeller flange.
14. Remove the D-5945 feedback collar retractor tool from the propeller.

***** END OF THE TASK *****



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PROPELLER INSTALLATION

Date : WO Number :
Part No. Propeller : 4HFR34C778- A/C Total Hours :
Serial No. Propeller : A/C Total :
Landing
Propeller Time : TSN: TSO:
Install to A/C Reg. :

Description	Eng.	RII	Remarks
B. INSTALL PROPELLER (Refer to Figure 01 to 04).			
NOTE: McCauley recommends that the propeller mounting nuts (McCauley part number A-1639-32) be replaced at each propeller installation, whenever possible. However, nuts may be reused if the locking material prevents turning of the nut on the stud by hand.			
1. Install the D-5945 feedback collar retractor tool on the propeller.			
2. Remove protective cover from the end of engine propeller flange.			
3. Make sure the flange is clean and free of nicks and burrs.			
4. Make mounting sure stud that the holes are engine clean, propeller dry, and flange, free of dowels, nicks and and burrs.			
5. Remove protective cover from the propeller hub mounting the flange.			
6. Make sure and that the propeller studs hub mounting flange, dowel pin holes, are clean mounting and, undamaged.,			
7. Make sure that a new O-ring is installed in the groove of the propeller hub mounting flange. Lubricate the O-ring with engine oil prior to installation of the propeller. NOTE: Refer to the Installation Parts For Turbine Engine Propellers, Table 1002 for the O-ring part number. NOTE: In the past, new propeller assemblies shipped from McCauley, the propeller hub/engine O-ring was installed in the O-ring groove of new propellers and hub assemblies. This practice has been discontinued. The O-ring is now included in the propeller unattached parts kit, which is included in the box with the propeller or hub assembly. Install the O-ring according to the assembly instructions in this Owner/Operator Manual.			
8. Use a propeller sling and hoist, or additional personnel, to position the propeller close to the engine propeller flange and align engine			



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flange dowel pins with the dowel pin holes on the propeller hub mounting flange. Rotate engine propeller mounting flange as required to align the dowel pin holes.

A. Hoist straps must be a minimum of 4 inches (100 mm) wide

B. The sling and hoist should have a weight limit rating at least twice the weight of the propeller that is to be installed.

C. The straps of the propeller sling should be placed on two of the propeller blades at least 6 inches (152 mm) outboard of the propeller hub. Make sure you protect the deice boots or anti-ice shoes from potential propeller sling abrasion damage, if installed.

CAUTION: Propeller must be installed straight onto the engine flange. Any rocking of the propeller with respect to the flange could result in damage to the engine/propeller flange mating surfaces.

9. Mount the propeller on the engine propeller shaft.

10. Make sure the alignment mark on the spinner aft bulkhead and the propeller blade with are in alignment.

11. Ensure threads of nuts and studs are free of burrs, nicks, and similar damage, and clean of foreign material.

CAUTION: Do not use oil as a substitute for approved lubricant. It is imperative that the correct specification of lubricant be used during installation. Substitution of the approved grease with an unapproved lubricant { or no lubricant} could result in undertorquing or severe over-torquing of propeller attaching parts.

12. Lubricate the threads of studs and nuts and the faces of nuts, spacers, or washers with MIL-PRF-83483 (McCauley part number A-1637-16) grease.

13. Install mounting nuts on mounting studs.

14. Torque the mounting nuts in an alternating sequence to prevent the hub rocking on the engine flange.

15. When the hub is seated fully on the engine flange, torque to the specification called out in the mounting decal located on propeller hub at the number 1 socket.

NOTE: If the decal containing the propeller installation instructions is missing or illegible, install a new decal. All Pratt & Whitney engine installations, use a part number A- 2230-7 decal. The A-2230-7 decal specifies a lubricated 68 to 72 foot-pounds (92.196 to 97.619 N-m) torque.

16. After you apply the final torque, apply torque seal to nut and stud threads.

17. If required, install the deice leads.

18. McCauley Torque Wrench Adapter:



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CAUTION: If an adaptor or extension (such as McCauley part number B-5588) is attached to torque wrench drive end and this adds to its length, then the actual applied torque will be greater than the dial reading. The following formula should be used to find what the dial should read in order to obtain the correct applied torque:

$$\text{Dial Reading} = \frac{\text{Torque Wrench Length} \times \text{Desired Torque}}{\text{Torque Wrench Length} + \text{Extension Length}}$$

19. Remove the D-5945 feedback collar retractor tool from the propeller

20. Make sure of proper rigging of engine controls. Refer to aircraft maintenance manual or STC maintenance manual supplement.

a. Feather, reverse, and low blade angles are set during assembly or overhaul. These angles are NOT adjustable in the field.

CAUTION: Do not operate the propeller below the minimum propeller idle speed operating restriction. The minimum propeller idle speed operating restriction is the result of a specific vibratory resonant condition known as "reactionless mode". Ground operation, at or near a reactionless mode vibratory resonance speed, can cause very high stresses in the propeller blades and hubs. These high stresses are more severe when operating in a tail-wind condition. If the propeller is operated within a restricted RPM range or below a minimum RPM restriction for an extended period of time, the propeller blades and hubs may become unairworthy due to fatigue. Hub or blade failure has the potential of causing a catastrophic event due to blade separation. The propeller RPM restriction is often placed below the minimum idle RPM; however, certain aircraft have a restriction that is above the propeller idle RPM setting. Either restriction is important. The propeller operating restrictions or limitations may be found in the Airplane Flight Manual (AFM) or Airplane Flight Manual Supplement (AFMS). The propeller installations may be controlled by the various airframe manufacturers Type Certificate (TC) or by Supplemental Type Certificate (STC).


21. Install Propeller Spinner

22. Start engine I.A.W Pilots Operating Handbook and FAA

23. Perform propeller dynamic balancing ref. C208B MM chapter 61-11-00 Dynamic balancing (McCauley) - Adjustment test. Refer also to related balancer tools manual.

24. Make an appropriate entry in Work Order and Aircraft Flight & Maintenance Log (AFML).

***** END OF THE TASK *****

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MAINTENANCE RELEASE

I hereby certify that the above stated maintenance and/or inspection was performed in accordance with the approved Aircraft Maintenance Manual and meets requirements of Civil Aviation Safety Regulations.


Name : _____

Stamp : _____

Signature : _____

Place/Date : _____

- END -

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PROPELLER CHANGE – Component Inventory Record			
Registration	:	Work Order Number	:
Airframe Time	:	Airframe Landing	:
Propeller Time	:	Propeller Cycle	:

Propeller OFF				Propeller ON		
Description	Part Number	Serial Number	Time Remaining	Part Number	Serial Number	Time Remaining
Propeller Hub						
Blade#1						
Blade#2						
Blade#3						
Blade#4						

NOTE: ANY OTHER COMPONENT CHANGES MUST BE FILLED ON ADDITIONAL WORKSHEET (SCA-MTC 030)



NON ROUTINE CARD
(Form: SCA/MTC/047)

1. JO/WO #	2. DATE	3. MTC TYPE	4. A/C REG/MSN
WO/002-SNV/IV/2023		REPLACEMENT	PK-SNv
5. CARD #	6. ATA SPEC	7. TRADE	8. STA
#001	61		
9. ZONE	10. PANEL		
FRONT			

11. DESCRIPTION			
PERFORM PROPELLER ASSY REPLACEMENT MODEL 4hFR34C778 REF EO NO. 002/EO/TEK-TS/IV/2023			
S/N ON:181192			
REFERENCE	<input checked="" type="checkbox"/> 002/EO/TEK-TS/IV/2023	<input checked="" type="checkbox"/> MCCAULEY MPC 700/750	<input type="checkbox"/> OTHER
RII (*)	<input checked="" type="checkbox"/> Y	<input type="checkbox"/> N	MHR :

12. RESULT			MECH	ENG	INSP (*)
Performed at A/C TT : A/C TC /LDG :					
FINDING	<input type="checkbox"/> Y	<input type="checkbox"/> N	ACT MHR :	DATE/TIME (DD/MM/YY)	
INSPECTION CARD (IC) #					

13. PARTS REQUIRED				
DESCRIPTION	PART NO	QTY	REMARK	
			STOCK	STATUS

14. TOOLS REQUIRED			
DESCRIPTION	PART NO / MODEL	NEXT CALIBRATION DATE	STATUS



Additional Work Sheet

Propeller Assy Replacement

Aircraft Registration: **PK-SNV**

WO# Nr: **WO/040-SNV/IV/2023**

Parts Used Sheet

Special Tool Used

[illegible]



Additional Work Sheet

Propeller Assy Replacement

Aircraft Registration: **PK-SNV**

WO# Nr: **WO/040-SNV/IV/2023**

Parts Used Sheet

Part Used

[illegible]



MAINTENANCE PROGRAM CESSNA 208/208B

Appendix E06.7 – OOP61001 / Propeller Dynamic Balance

Reg. Mark : PK - _____ Date : _____
MSN : _____ Station : _____
TSN / CSN : _____ WO No. : _____

NO.	ZONE	TASK	SIGNATURE	
			ENGINEER SIGN&STAMP	RII SIGN&STAMP
01	211 212	Perform propeller dynamic balancing refer to Cessna Maintenance Manual 61-11-00.		
*** End of OOP61001 Items ***				

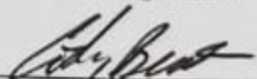
PERSONNEL PARTICIPATING IN THIS INSPECTION			
NAME	POSITION	SIGNATURE	LICENSE NUMBER

RETURN TO SERVICE

The work recorded above has been carried out in accordance with the requirements of the Civil Aviation Safety Regulation for the time being in force and in that respect the aircraft is consider fit for Release to Service.

Name : _____ Place/Date : _____

Sign & Stamp : _____

1. Approving Civil Aviation Authority/Country: FAA/United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: YA-08102022-CB-01	
4. Organization Name and Address: YINGLING AVIATION, 2010 AIRPORT RD. WICHITA, KS. 67209 316-943-3246 YN8R621Y					5. Work Order/Contract/Invoice Number: 25460	
6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:	
1	PROPELLER	P7785550-01	1	181192	OVERHAUL	
12. Remarks: Overhauled propeller. All work accomplished in accordance with McCauley Maintenance Manuals: MPC750 Series Overhaul Manual Rev.4, BOM100 Rev.11, and SPM100 Rev.8. For detailed work instruction see log entry dated 08/10/2022, or WO#25460. PROPELLER ASSEMBLY TT: 90.4 TSO: 0.0 BLADE S/N #1 AOK30032 BLADE S/N #2 ANA30036 BLADE S/N #3 ANA30039 BLADE S/S #4 ANA30046						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature: XXXXXXXXXXXXXXXXXXXXXXXXXXXX		13c. Approval/Authorization No.: XXXXXXXXXXXXXXXXXXXX		14b. Authorized Signature: 		14c. Approval/Certificate No.: FAA CRS YN8R621Y
13d. Name (Typed or Printed): XXXXXXXXXXXXXXXXXXXXXXXXXXXX		13e. Date (dd/mm/yyyy): XXXXXXXXXXXXXXXXXXXX		14d. Name (Typed or Printed): CODY BEAT		14e. Date (dd/mm/yyyy): 10/AUG/2022
User/Installer Responsibilities						
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.</p> <p>Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.</p> <p>Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>						