



PT. SMART CAKRAWALA AVIATION

**WORK ORDER**

Form: SCA/MTC/030

Subject :	No.	WO/050-SNK/IX/2023
<b>Engine Replacement</b>	Date	8 Sept 2023
	A/C Reg.	PK-SNK C208-00658
Reference :	Prepared By	TS
MP C208B Issued 01	Checked By	CI
	Approved By	TM

To : Engineer In Charge

**Description :**

1. Perform Engine Replacement
2. Make an entry in Maintenance Log.
3. Return the Completed Work Order and Form to PPC.

#If any finding, please close the routine card, and transferred to inspection card.

**Additional Work :**

Compliance Statement	Sign & Date Company Lic. No.:  (Engineer In Charge)	Signature  (Technical Manager)
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**AIRCRAFT CHECK WORK SUMMARY**  
(Form: SCA/MTC/051)

DATE OF ISSUED	JO/WO #	TYPE OF MAINTENANCE	DATE OF ACCOMPLISHED		
8 Sep 2023	WO/050-SNK/IX/2023	Replacement			
<b>AIRCRAFT DATA</b>					
Subject	Pos #	Serial Number (SN)	TTSN/TCSN		
Engine	#1	PCE-PC1288			
	#2	-			
Propeller/Rotor	#1	190345			
	#2	-			
Landing Gear	NLG				
	LH MLG				
	RH MLG				
<b>PACKAGE COVERED</b>					
No	Subject		Qty	Remark	
1	Non-Routine Card		1		
2	Inspection Card		1		
3	Work Order		1		
4	Summary Inspection List		1		
5	Material and Tool List		-		
6	Escalation form		-		
7	CRS (SMI / Unscheduled Maintenance)		1		
<b>INSPECTION CARD (IC) LIST (Finding during maintenance)</b>					
No	Taskcard Ref	Subject	Status		Name/ Sign & Stamp
			Open	Close	
<u>IC-001</u>					
<u>IC-002</u>					
<u>IC-003</u>					
<u>IC-004</u>					
<u>IC-005</u>					
<u>IC-006</u>					

<u>IC-007</u>					
<u>IC-008</u>					
<u>IC-009</u>					
<u>IC-010</u>					
<u>IC-011</u>					
<u>IC-012</u>					
<u>IC-013</u>					
<u>IC-014</u>					
<u>IC-015</u>					

Prepared by :  
Technical Support



Hani

Checked by :  
Chief Maintenance



Dodit

Verified by :  
Chief Inspector



Yanuar

Approved by :  
Technical Manager



Istiono



PT. SMART CAKRAWALA AVIATION

**CERTIFICATE RETURN TO SERVICE**  
SCHEDULED MAINTENANCE INSPECTION  
(CRS-SMI)

A/C TYPE	: CESSNA 208		TTSN	:
A/C REG	: PK-SNK		TCSN	:
MSN	: C208-00658		DATE	:
TYPE OF INSPECTION	: ENGINE REPLACEMENT			
DUE AT	: 3600 HRS			
REFF	: MP C208B ISSUED 01			
EXCEPTION				
<b>AUTHORIZED PERSON</b>				
I hereby certify that this aircraft has been maintained accordance with CASR and Maintenance Program. Aircraft safe and airworthy for flight				
NAME	CAT	AMEL/OTR NO	SIGN&STAMP	DATE
	AIRFRAME & POWER PLANT			
	EIRA			
THE NEXT DUE TYPE OF INSPECTION	:			
DUE AT	:			
<b>Form: SCA/MTC/049</b>				



**SUMMARY INSPECTION ITEMS**  
**(Form: SCA/MTC/050)**

WO Ref: WO/050-SNK/IX/2023

NO.	TASK CARD NO.	DESCRIPTION	DATE	EST MHR	NAME	STAMP
1	NRC-001	ENGINE REPLACEMENT / EO-011				

	<b>INSPECTION CARD</b> (Form: SCA/MTC/ 048)				TECHNICAL DEPARTMENT			
1. CARD #	2. JO/WO #	3. ORIGINATOR	4. CARD REF	5. DATE				
6. A/C REG/MSN	7. A/C TYPE	8. TRADE	12. VENDOR ORDER #					
9. ZONE	10. STA	11. MTC TYPE						
13. DESCRIPTION/DEFECT-IF FINDING OF CPCP INSPECTION, PLEASE COMPLETE SET. 20						14 PPC/ENG	15 DATE	
16. CORRECTIVE ACTION						17 MECH	18 ENG. LIC	19 DATE
Performed at A/C TT : ..... A/C TC /LDG : .....								
20. CORROSION INFORMATION								
LOCATION			CAUSE OF DAMAGE					
			<input type="checkbox"/> Environment					
			<input type="checkbox"/> Internal Leakage					
CORROSION <input type="checkbox"/> Isolated <input type="checkbox"/> Widespread			<input type="checkbox"/> Chemical Spill					
CORROSION LVL <input type="checkbox"/> 1 <input type="checkbox"/> 2 <input type="checkbox"/> 3			<input type="checkbox"/> LAV/Galley Spill					
PROPOSED ACTION <input type="checkbox"/> Doublers			<input type="checkbox"/> Blocked Drain					
<input type="checkbox"/> Others			<input type="checkbox"/> Wet Insulation Blanket					
.....			<input type="checkbox"/> Other					
21. If the defect is RII, Please Sign this card finally by RII Inspector						INSP	DATE	
NOTICE OF INSPECTOR								
22. PARTS REQUIRED								
PART DESCRIPTION		PART NO		QTY	SERIAL NO		STATUS	
				ON	OFF		CLOSE	OPEN
23. TOOLS REQUIRED								
DESCRIPTION		PART NO. / MODEL		NEXT CALIBRATION DATE			STATUS	



NRC No. : 001

**NON ROUTINE CARD**  
(Form: SCA/MTC/047)

1. JO/WO #	2. DATE	3. MTC TYPE	4. A/C REG/MSN
WO/050-SNK/IX/2023		COMPONENT REPLACEMENT	PK-SNK
5. CARD #	6. ATA SPEC	7. TRADE	8. STA
001	72		
9. ZONE	10. PANEL		
ENGINE			

## 11. DESCRIPTION

## PERFORM ENGINE REPLACEMENT

P/N OFF: PT6A-114A / 3044000 P/N ON: PT6A-114A / 3044000

S/N OFF: PCE-PC 1988 S/N ON: PCE-PC 1288

REFERENCE	<input checked="" type="checkbox"/> 011/EO/TEK-TS/IX/2023	<input type="checkbox"/>	<input type="checkbox"/> OTHER
RII (*)	<input type="checkbox"/> Y	<input type="checkbox"/> N	MHR :

12. RESULT				MECH	ENG	INSP (*)
Performed at A/C TT : ..... A/C TC /LDG : .....						
FINDING	<input type="checkbox"/> Y	<input type="checkbox"/> N	ACT MHR :	DATE/TIME (DD/MM/YY)		
INSPECTION CARD (IC) #						

## 1. PARTS REQUIRED

DESCRIPTION	PART NO	QTY	REMARK	
			STOCK	STATUS

## 1. TOOLS REQUIRED

DESCRIPTION	PART NO / MODEL	NEXT CALIBRATION DATE	STATUS

Distribution : 1. White : PPC/Engineering

2. Red : Quality

3. Yellow : Retain on Log Book



TECHNICAL SUPPORT  
TECHNICAL DEPARTMENT  
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**ENGINEERING ORDER**

**011/EO/TEK-TS/IX/2023**

**REMOVAL & INSTALLATION OF ENGINE ASSY  
PT6A-114A ON CESSNA C208B**

**PT. SMART CAKRAWALA AVIATION**

Prepared	Checked	Approved
<b>Technical Support</b>	<b>Chief Inspector</b>	<b>Technical Manager</b>
Signature: 	Signature: 	Signature: 
Name: Dwi M.	Name: Yanuar A. F.	Name: Istiono
Date: 08 Sep 2023	Date: 08 Sep 2023	Date: 08 Sep 2023



TECHNICAL SUPPORT  
TECHNICAL DEPARTMENT  
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## SMART AVIATION ENGINEERING ORDER

No. EI:  
**011/EO/TEK-TS/IX/2023**

Rev. No. :  
**Original**

Date Issued :

**08 September 2023**

Task Description :

**REMOVAL & INSTALLATION OF ENGINE  
ASSY PT6A-114A ON CESSNA C208B**

Data Reference :

- Model 208 Series Maintenance Manual  
Revision 37, Revision Date Mar 1, 2020  
Chapter 71 Power Plant – Maintenance  
Practices**

Aircraft Type :

**CESSNA C208B WITH ENGINE MODEL PT6A-  
114A / PT6A114**



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## SMART AVIATION ENGINEERING ORDER

### 1. Description.

This EO is issued, to perform removal & installation checklist powerplant maintenance practices the PT6A-114/PT6A-114A engine on Cessna C208B.

### 2. Aircraft Effectivity.

REGISTRATION	SERIAL NUMBER
PK-SNK	208-00658

### 3. Compliance

The Engine model PT6A-114A have TBO 3600 Hours, do a removal the engine installed on airframe refer to accomplishment instruction task card, and install the overhauled/new engine on the aircraft refer to accomplishment instruction task card.

### 4. Distribution.

TECHNICAL MANAGER	[ <input type="checkbox"/> ]	MATERIAL SUPPORT	[ <input type="checkbox"/> ]
SAFETY & QUALITY MANAGER	[ <input type="checkbox"/> ]	TECHNICAL SUPPORT	[ <input type="checkbox"/> ]
CHIEF INSPECTOR	[ <input type="checkbox"/> ]	FILE	[ <input type="checkbox"/> ]

### 5. Manhours

32.0 man-hour to do the inspection.

### 6. Material.

PWC09-005	Compound, Universal
PWC09-006	Compound, Universal
PWC11-027	Solvent, Petroelum
PWC11-031	Cleaner, Engine

### 7. Special Tool Required.

PWC34300	Stand, Engine
PWC51861-600	Sling Assembly, Engine

### 8. Publication Affected.

None.



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## SMART AVIATION ENGINEERING ORDER

### 9. Accomplishment Instructions.

#### C208B ENGINE REMOVAL

Date : _____	Work Number : _____
Part No. Engine : PT6A-114A	A/C Total Hours : 4226:11 Hrs _____
Ser. No. Engine : PCE-PC 1988	A/C Total Landings : 5983 Ldg _____
Engine Time TSN: 7062:36 Hrs	TSO: 3762:13 Hrs
CSN: 106236 Cyc CSO: 7923 Cyc	
Removed from A/C Reg. : PK-SNK	

Description	Eng.	RII	Remarks
<b>A. REMOVE ENGINE (Refer to Figure 01 and Figure 02)</b>			
<b>CAUTION:</b> Chock main wheels and place a tailstand under tailcone before attempting engine removal.			
1. Turn electrical power off.			
2. Pull fuel firewall shutoff control out (off).			
3. Remove upper cowling doors and lower cowling panels.			
4. Drain residual fuel from lines and fuel filter using filter drain. Remove fuel supply hose at fuel heater. Remove fuel motive flow hose at fuel control unit.			
5. Remove right nose cap and oil cooler.			
6. Remove top cowl center panel assembly and nose cap.			
7. Remove propeller.			



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8. Disconnect and remove propeller speed control cable.			
9. Remove the left nose cap/induction air duct/inertial air separator.			
10. Disconnect cabin heater bleed air line at flow control valve and bleed air hose at mixing air valve.			
11. Remove starter/generator cooling air hose from starter/generator.			
12. Remove engine fire detector wiring harness.			
13. Disconnect electrical wiring connectors and ground wires at the following equipment locations: i) Propeller overspeed governor and ITT harness (left front of engine).			
ii) Propeller tachometer generator (right front of engine)			
iii) Cabin bleed air heater flow control valve (lower right side of engine).			
iv) Oil temperature sensor (right side of engine).			
v) Fuel control heater (right rear of engine).			
vi) Gas generator section tachometer generator (lower right side of engine).			
vii) Starter/generator (center top of engine accessory case).			
viii) Ignition exciter high tension leads at ignition exciter (right engine mount truss).			



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14. Disconnect engine power control cables at fuel control unit.			
15. Remove torquemeter pressure and vent lines at forward upper right side of engine mount truss.			
16. Connect hoist sling to forward and aft lifting brackets and connect sling to engine hoist.			
17. Raise hoist to just support weight of engine and remove nuts and bolts at each of four corners of engine mounting ring.			
18. Ensure all wiring and lines are free, then carefully move hoist and engine forward to clear engine mount truss.			
19. If engine is to be returned for overhaul or replaced, remove the following items:  i) Engine induction air plenum. Refer to Chapter 71, Engine cowling and Nose Cap - Maintenance Practices.			
ii) Engine mount ring, elastomers, and engine mount brackets. Refer to Chapter 71, Engine mount - Maintenance Practices.			
iii) Propeller overspeed governor. Refer to Chapter 61, Propeller Control - Maintenance Practices.			
iv) Propeller tachometer generator. Refer to Chapter 77, Propeller RPM Indicator - Maintenance Practices.			
v) Oil temperature sensing sensor. Refer to Chapter 79, Oil Indicating - Maintenance Practices.			
vi) Oil cooler bracket and pressure/return hoses. Refer to Chapter 79, Oil Distribution - Maintenance Practices.			



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vii) Standby alternator (if equipped). Refer to Chapter 24, Standby Electrical System - Maintenance Practices.

viii) Torque sensing line and fittings.

20. Make and inventory record P/N and S/N of the engine and its accessories from the engine that removed, fill out into the List (Form Engine Change – Major Component Inventory Record)

21. Make an appropriate entry in Work Order (WO) and Aircraft Flight Maintenance Log (AFML)

**\*\*\* END OF THE TASK \*\*\***



TECHNICAL SUPPORT  
TECHNICAL DEPARTMENT  
**ENGINEERING ORDER**

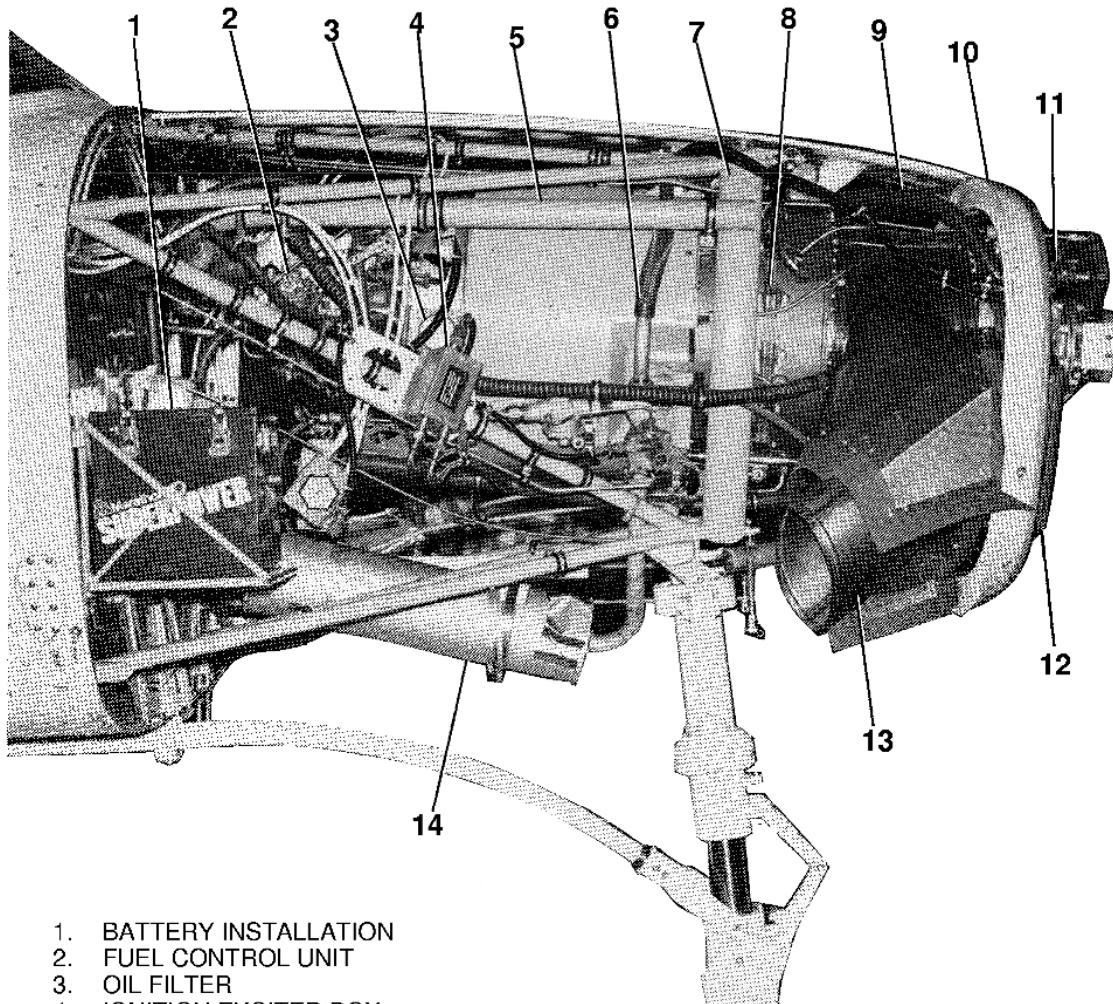
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## SMART AVIATION ENGINEERING ORDER

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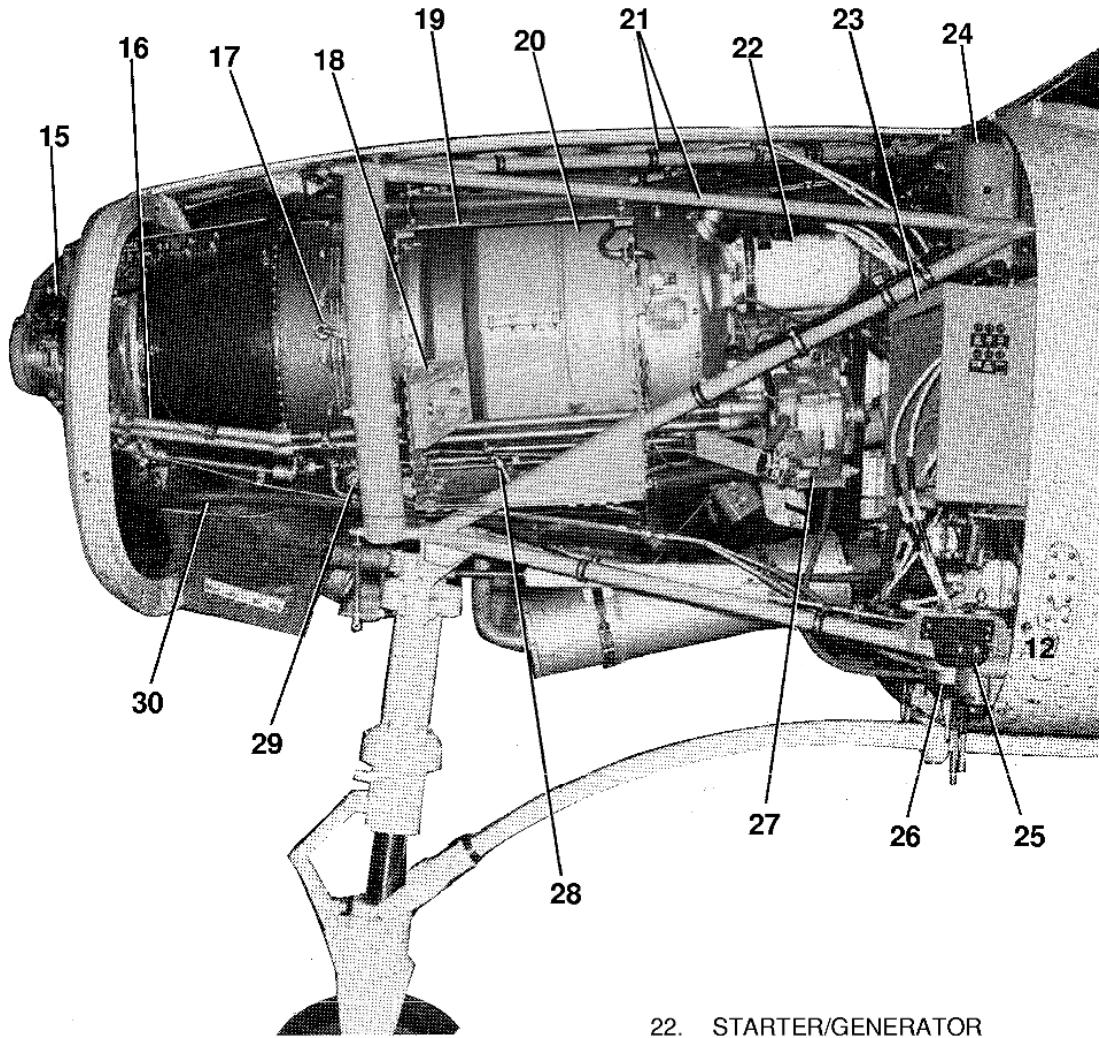
1. BATTERY INSTALLATION
2. FUEL CONTROL UNIT
3. OIL FILTER
4. IGNITION EXCITER BOX
5. STARTER/GENERATOR COOLING AIR BLAST TUBE
6. BLEED AIR PRESSURE LINE
7. ENGINE MOUNT RING
8. FUEL MANIFOLD
9. OIL RETURN FROM OIL COOLER
10. RIGHT COWLING BULKHEAD
11. PROPELLER GOVERNOR
12. OIL COOLER
13. PRIMARY EXHAUST STACK
14. BLEED AIR HEATER MUFFLER

2650X1002

**Figure 01 Sheet 1**

## SMART AVIATION ENGINEERING ORDER

A21759



15. PROPELLER OVERSPEED GOVERNOR	22. STARTER/GENERATOR
16. REDUCTION GEARBOX	23. POWER DISTRIBUTION BOX
OIL LINES	24. STANDBY ALTERNATOR
17. SPARK IGNITER	CONTROL UNIT
18. ENGINE MOUNT BRACKET	25. AUXILIARY POWER RECEPTACLE
19. INDUCTION AIR PLENUM	26. FUEL FILTER
20. COMPRESSOR INLET	27. STANDBY ALTERNATOR
21. ENGINE MOUNT TRUSS	28. COMPRESSOR DRAIN LINE
	29. FUEL MANIFOLD DUMP VALVE
	30. OIL COOLER PRESSURE HOSE

2650X1003

**Figure 1 Sheet 2**



TECHNICAL SUPPORT  
TECHNICAL DEPARTMENT  
**ENGINEERING ORDER**

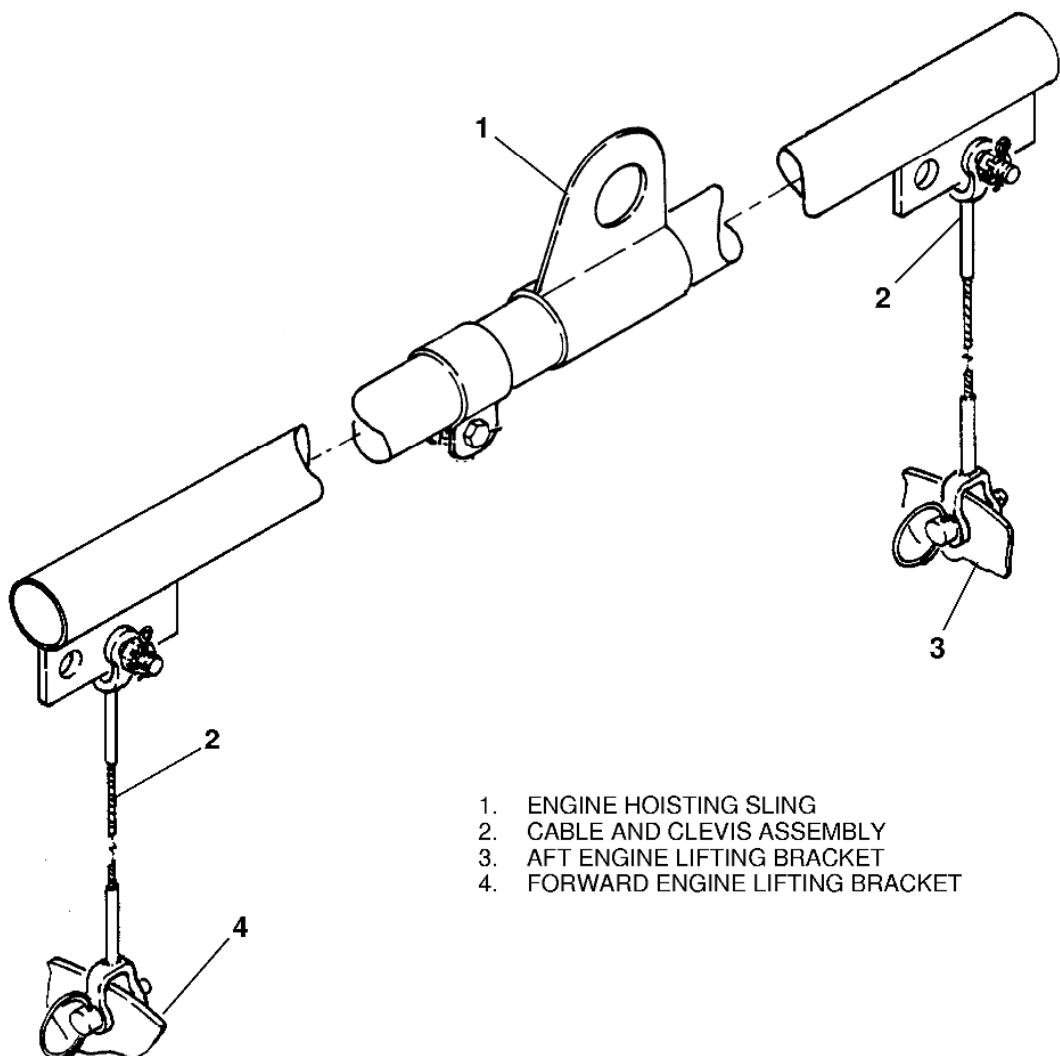
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## SMART AVIATION ENGINEERING ORDER

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2680X1044

**Figure 02**



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## SMART AVIATION ENGINEERING ORDER

### C208B ENGINE INSTALLATION

Date : \_\_\_\_\_ Work Number : \_\_\_\_\_  
Part No. Engine : PT6A-114A A/C Total Hours : \_\_\_\_\_  
Ser. No. Engine : PCE-PC 1288 A/C Total Landings : \_\_\_\_\_  
Engine Time TSN: 6989:64 TSO: 0:00  
CSN: 9238 Cyc CSO:0  
Installed on A/C Reg. : PK-SNK

Description	Eng.	RII	Remarks
<b>B. INSTALL ENGINE (Refer to Figure 01 and Figure 02).</b>			
1. Make and inventory record P/N and S/N of the engine and its accessories. Fill out into the List (Form Engine Change – Major Component Inventory Record)			
2. Install engine mount brackets, elastomers, and engine mount ring. Refer to Chapter 71, Engine mount – Maintenance Practices.			
3. Connect lifting hoist sling to forward and aft lifting brackets on engine and lift engine into position forward of engine mount truss.			
4. Make sure that all engine lines and equipment are clear.			
5. Lubricate the engine mount bolts with MIL-PRF-81322G Grease, before you install them to prevent corrosion.			
6. Make sure that the threads of bolts are covered during application of grease. Lubrication on threads can alter the torque reading.			
7. Move the hoist and engine aft to align the engine mount ring holes with the holes in the engine mount truss.			
8. Install the mount bolts (engine mount truss to engine mount ring) and torque the bolt/nuts to 450 to 500 inch-pounds (50.8 to 56.4 N·m). Remove the hoist and sling.			



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9. Connect torque meter pressure and vent lines at upper left firewall. Bleed torque meter indicating system.			
10. Connect engine power controls at fuel control unit. Rig controls.			
11. Connect the electrical leads of the following items of electrical equipment:  i) Ignition exciter high tension leads at ignition exciter (right engine mount truss).  ii) Starter/generator (center top of engine accessory case).  iii) Gas generator section tachometer generator (lower right side of engine).  iv) Fuel control heater (right rear of engine).  v) Oil temperature sensor (right rear of engine).  vi) Cabin bleed air heater flow control valve (lower right side of engine).  vii) All engine to engine mount ground straps.  viii) Propeller overspeed governor and ITT harness (left front of engine).  ix) Propeller tachometer generator (right front of engine).			
12. Install engine fire detector warning harness.			
13. Connect starter/generator cooling air hose to starter/generator.			



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14. Connect engine bleed air line to cabin bleed air heater flow control valve. Connect engine bleed air hose to cabin bleed air heater mixing air valve.			
15. Install left nose cap/induction air duct/inertial air separator, if not previously installed.			
16. Install propeller, if not previously installed.			
17. Install and connect propeller governor control cable.			
18. Install left and right nose cap bulkhead assemblies and top cowling center panel.			
19. Install oil cooler and right nose cap.			
20. Connect fuel supply hose at fuel heater and fuel motive flow hose at fuel control unit.			
21. Push fuel firewall shutoff control fully in.			
22. With fuel line disconnected at fuel manifold below engine, motor engine with starter to purge fuel lines.			
23. Perform RII Dual Inspection <b>before</b> to first engine start.			
24. Start engine and perform operational check. Refer to Pilot's Operating Handbook and FAA-Approved Airplane Flight Manual.			
25. Perform Ground Run, Use the Pratt and Whitney PT6A-114/-114A/-135/-135A Engine Maintenance Manual with the Pilot's Operating Handbook and FAA-Approved Airplane Flight Manual to do the operational check of the different components on the engine.			
26. Shut down engine and check for fluid leaks, connections or hardware, etc.			
27. Perform RII inspection if any controls have been disturbed or adjusted.			



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28. Install engine cowling.

29. Make an appropriate entry in Work Order (WO) and  
Aircraft Flight Maintenance Log (AFML)

### MAINTENANCE RELEASE

I hereby certify that the above stated maintenance and/or inspection was performed in accordance with the approved Aircraft Maintenance Program and meets requirements of Civil Aviation Safety Regulations.

Name : \_\_\_\_\_ Stamp : \_\_\_\_\_

Signature : \_\_\_\_\_ Place/Date : \_\_\_\_\_



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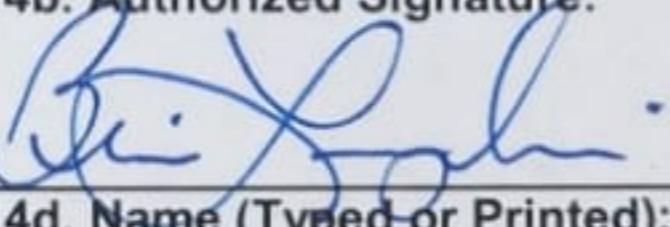
Rev. Date      08/09/2023

**ENGINE CHANGE - Major Component Inventory Record**

<b>Registration</b>	:	<b>Work Order Number</b>	:
<b>Airframe Time</b>	:	<b>Airframe Landings</b>	:
<b>Engine Time</b>	:	<b>Engine Cycle</b>	:

<b>Description</b>	<b>Engine OFF</b>			<b>Engine ON</b>		
	<b>Part Number</b>	<b>Serial Number</b>	<b>Time Remaining</b>	<b>Part Number</b>	<b>Serial Number</b>	<b>Time Remaining</b>
Engine Assembly						
Propeller Assembly						
Compressor Bleed Valve						
Fuel Control Unit						
Oil Fuel Heater						
Igniter Exciter						
Flow Divider						
Oil Cooler						
Starter Generator						
Alternator						
Fuel Pump						
Propeller Governor						
Propeller Overspeed Governor						
Fuel Nozzle						

**NOTE: ANY OTHER COMPONENT CHANGES MUST BE FILL ON INSPECTION CARD (SCA/MTC/048)**

1. Approving National Aviation Authority/Country:  FAA/UNITED STATES	2. <b>AUTHORIZED RELEASE CERTIFICATE</b> FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG				3. Form Tracking Number:  PHX-080723-102938
4. Organization Name and Address:   <b>PRIMETURBINES</b>	<b>Prime Turbines LLC</b> 3130 North Oakland, Ste 104 Mesa, AZ 85215			<b>FAA Certificate 5TPR585C</b> Phone: 480-428-6341 Fax: 480-219-3587	5. Work Order/Contract/Invoice Number:  M5O1090
6. Item: 1	7. Description: ENGINE PT6A-114A	8. Part Number: 3044000	9. Quantity: 1	10. Serial/Batch Number: PCE-PC1288	11. Status/Work: OVERHAULED
<b>12. Remarks:</b> OVERHAULED in accordance to Pratt & Whitney Canada Overhaul Manual 3021243 Rev 50 dated 10/24/2022. SHP: 675 NG: 36200 SFC: 0.643 TRIM STICK P/N: 3031417CL50 725: ITT: 1760 CT VANE P/N: 3123001CL01 S/N HLA600P CAV: 5S2 EFA: 6.13 PT VANE P/N: 3024682CL13.1 S/N 9M739 CAV: 5S6 EFA: 15.98					
Details regarding the work performed are on file at Prime Turbines, LLC. under the W/O number listed in block (5)					
TSN: 6,989.64 TSO: 0.0 CSN: 9,238 CSO: 0.0 This repair station certifies that the work specified in Block 11/12 was carried out in accordance with EASA Part 145 and in respect to that work, the article is considered ready for release to service under the EASA Part 145 Approval Number. EASA 145.6705					
13a. Certifies the items identified above were manufactured in conformity to:			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12  <input type="checkbox"/> Approved design data and are in condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.		
13b. Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature: 		14c. Approval/Certificate No.:  5TPR585C
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): BRIAN LANGEHEINE		14e. Date (dd/mmm/yyyy): 02/AUG/2023

### User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: FAA/United States	2. <b>AUTHORIZED RELEASE CERTIFICATE</b> FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: 52196			
4. Organization Name and Address:  Heritage Turbines, Inc. Repair Station # H1IR069Y 35 Hinckley Road Barnstable Municipal Airport Hvannis, MA 02601 1-508-778-7788		5. Work Order/Contract/Invoice Number: 012076			
6. Item: 1	7. Description: Bleed Off Valve	8. Part Number: 3049038-03 (540-1407-4)	9. Quantity: 1	10. Serial Number: 4963	11. Status/Work: Overhauled

**12. Remarks:**

Overhauled Bleed Valve Assy I/A/W Honeywell Component Maintenance Manual P/N 540-1407 Rev.2 Dated Apr. 12, 2017. ATA chapter reference 75-31-44  
Control Pressure set at 31.074 PSIA at 70 PSIA I/A/W P&W S.B. 1588R2.

ESN: PCE-PC1288

Certifies that the work specified in block 11/12 was carried out in accordance with EASA Part-145 and in respect to that work the component is considered ready for release to service under EASA Part-145 Approval Number: EASA.145.6342

13a. Certifies the items identified above were manufactured in conformity to:  <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12  Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature:	14c. Approval/Certificate No.: H1IR069Y
13d. Name (Typed or printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Jesse Johnson	14e. Date (dd/mmm/yyyy): 06/Jun/2023

**User/Installer Responsibilities**

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

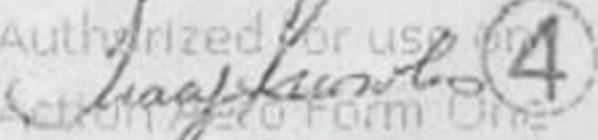
1. Approving Civil Aviation Authority/Country:  Transport Canada/Canada	2. <b>AUTHORIZED RELEASE CERTIFICATE</b> FORM ONE	3. Form Tracking Number:  <b>W119291</b>			
4. Organization Name and Address: Action Aero 91 Watts Avenue Charlottetown, PE, Canada C1E 2B7 Ph: 902-370-3311 Fax: 902-370-3313 Email: info@actionaero.com		5. Work Order/Contract/Invoice Number:  <b>012073</b>			
6. Item: 1	7. Description: FLOW DIVIDER & DUMP VALVE	8. Part Number: 3019906/25536-4	9. Quantity: 1	10. Serial Number: 9959512673	11. Status/Work: OVERHAULED

## 12. Remarks:

Overhauled IAW Triumph CMM 73-10-01 (3020539), Revision 7, Dated July 30/2004.  
TSO 0.0 Hours.

ESN:PCE-PC1288

The work specified in blocks 11 and 12 was carried out in accordance with EASA Part 145 and with respect to that work, the aircraft component(s) is/are considered ready for release to service under EASA approval number EASA.145.7199.

13a. Certified the items identified above were manufactured in conformity to:  <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12	14a. <input checked="" type="checkbox"/> CAR 571.10 Maintenance Release <input checked="" type="checkbox"/> Other regulation specified in Block 12  Certifies that, except where otherwise specified in block 12, the work identified in block 11 and described in block 12 has been performed in compliance with the Canadian Aviation Regulations.		
13b. Authorized Signature:	13c. Approval Authority/Certification No.:	14b. Authorized Signature:  	14c. Approval/Certificate No.:  AMO# 1-08
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed):  Tracy Knowles	14e. Date (dd/mmm/yyyy):  29-May-2023
User/Installer Responsibilities			

(Previously Form 24-0078)

This certificate does not constitute authority to install.

Installers working in accordance with the national regulations of a country other than that specified in block 1, must ensure that their regulations recognize certifications from the country specified.

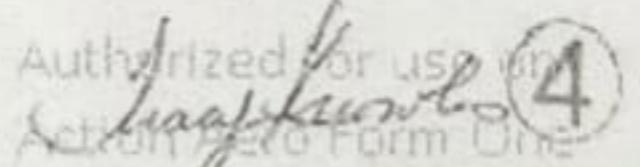
Statements in blocks 13A or 14A do not constitute installation certification. In all cases, the technical record for the aircraft must contain an installation certification, issued in accordance with the applicable national regulations before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country: <b>Transport Canada/Canada</b>	2.	<b>AUTHORIZED RELEASE CERTIFICATE</b> <b>FORM ONE</b>			3. Form Tracking Number: <b>W119285</b>
4. Organization Name and Address: <b>Action Aero</b> <b>91 Watts Avenue</b> <b>Charlottetown, PE, Canada C1E 2B7</b> <b>Ph: 902-370-3311</b> <b>Fax: 902-370-3313</b> <b>Email: info@actionaero.com</b>					5. Work Order/Contract/Invoice Number: <b>012074</b>
6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:
1	MAIN ENGINE FUEL PUMP	702801-5	1	001251	OVERHAULED

## 12. Remarks:

Argo-Tech Fuel Pump Installation Number 702800 P&WC PN 3034792. ESN: PCE-PC1288  
Overhauled IAW Argo-Tech CMM 73-10-11, Revision 3, Dated October 01/2007. TSO: 0.0 hours.

The work specified in blocks 11 and 12 was carried out in accordance with EASA Part 145 and with respect to that work, the aircraft component(s) is/are considered ready for release to service under EASA approval number EASA.145.7199.

13a. Certified the items identified above were manufactured in conformity to:	14a. <input checked="" type="checkbox"/> CAR 571.10 Maintenance Release <input type="checkbox"/> Other regulation specified in Block 12		
<input type="checkbox"/> Approved design data and are in a condition for safe operation.	Certifies that, except where otherwise specified in block 12, the work identified in block 11 and described in block 12 has been performed in compliance with the Canadian Aviation Regulations.		
13b. Authorized Signature:	13c. Approval Authorization No.:	14b. Authorized Signature: 	14c. Approval/Certificate No.: AMO# 1-08
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed): Tracy Knowles	14e. Date (dd/mmm/yyyy): 21-Jun-2023

(Previously Form 24-0078)

**User/Installer Responsibilities**

This certificate does not constitute authority to install.

Installers working in accordance with the national regulations of a country other than that specified in block 1, must ensure that their regulations recognize certifications from the country specified.

Statements in blocks 13A or 14A do not constitute installation certification. In all cases, the technical record for the aircraft must contain an installation certification, issued in accordance with the applicable national regulations before the aircraft may be flown.

1. Approving Civil Aviation Authority/Country:  FAA/United States	2.	<b>AUTHORIZED RELEASE CERTIFICATE</b> FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number:  230908
4. Organization Name and Address:  Keystone Turbine Services, 885 Fox Chase, Suite 111, Coatesville, PA 19320 - Certificate No. 8MHR895B			5. Work Order/Contract/Invoice Number:  85018901-EA-FC		
6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:
1	FUEL CONTROL	3244897-4	1	C66799	OVERHAULED

12. Remarks:

OVERHAULED & TESTED IAW HONEYWELL CMM. 73-20-78 REV. 2 6/8/2021  
TSO: 0.0 TRACE TO ESN: PCE-PC1288

Certifies that the work specified in Block 11/12 was carried out in accordance with EASA Part 145 and in respect to that work the article is considered ready for release to service under EASA Part 145 approval no. EASA 145-6410.

13a. Certifies the items identified above were manufactured in conformity to:  <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 12  Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature:	14c. Approval/Certificate No.:
13d. Name (Typed or Printed):	13e. Date (dd/mm/yy):	14d. Name (Typed or Printed):	14e. Date (dd/mm/yy):

**User/Installer Responsibilities**

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

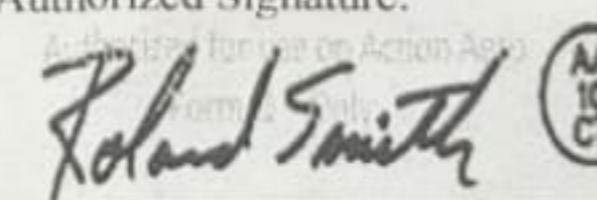
1. Approving Civil Aviation Authority/Country:  Transport Canada/Canada	2. <b>AUTHORIZED RELEASE CERTIFICATE</b> FORM ONE	3. Form Tracking Number:  <b>W119287</b>			
4. Organization Name and Address: <b>Action Aero</b> <b>91 Watts Avenue</b> <b>Charlottetown, PE, Canada C1E 2B7</b> <b>Ph: 902-370-3311</b> <b>Fax: 902-370-3313</b> <b>Email: info@actionaero.com</b>		5. Work Order/Contract/Invoice Number:  <b>012075</b>			
6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:
1	TURBOPROP GOVERNOR ASSEMBLY	8210-002-01	1	13039147	OVERHAULED

## 12. Remarks:

8210-002-01, Revision: NEW.  
Overhauled IAW Woodward Governor CMM 61-20-14, Revision 3, Dated September 30/1997.  
TSO 0.0 Hours.  
Complied with SB 33531C.

ESN: PCE-PC1288

The work specified in blocks 11 and 12 was carried out in accordance with EASA Part 145 and with respect to that work, the aircraft component(s) is/are considered ready for release to service under EASA approval number EASA.145.7199.

13a. Certified the items identified above were manufactured in conformity to:		14a. <input checked="" type="checkbox"/> CAR 571.10 Maintenance Release <input checked="" type="checkbox"/> Other regulation specified in Block 12	
<input type="checkbox"/> Approved design data and are in a condition for safe operation.  <input type="checkbox"/> Non-approved design data specified in Block 12		Certificates that, except where otherwise specified in block 12, the work identified in block 11 and described in block 12 has been performed in compliance with the Canadian Aviation Regulations.	
13b. Authorized Signature:	13c. Approval Authorization No.:	14b. Authorized Signature:  Roland Smith	14c. Approval/Certificate No.:  AMO# 1-08
13d. Name (Typed or Printed):	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed):  Roland Smith	14e. Date (dd/mmm/yyyy):  30-May-2023

(Previously Form 24-0078)

**User/Installer Responsibilities**

This certificate does not constitute authority to install.

Installers working in accordance with the national regulations of a country other than that specified in block 1, must ensure that their regulations recognize certifications from the country specified.

Statements in blocks 13A or 14A do not constitute installation certification. In all cases, the technical record for the aircraft must contain an installation certification, issued in accordance with the applicable national regulations before the aircraft may be flown.

1. Approving Civil Aviation  
Authority/Country:

FAA/UNITED STATES

2.

## AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

3. Form Tracking Number:

MRO4242

4. Organization  
Name and Address: AVIATION CONTROLS INC.  
d/b/a PRECISION AVIATION CONTROLS  
101 FREEDOM DRIVE  
INDEPENDENCE, KS 67301  
UNITED STATES

FAA Certificate : YNBR921L

5. Work Order/Contract/Invoice Number:

9209881

6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:
1	FUEL HEATER PT6A	10552E 3032710	1 EA	WA23408	OVERHAULED

12. Remarks: Referencing manual No. 73-10-03 Revision: 16 Revision Date: DEC 15 2016  
OVERHAULED IN ACCORDANCE WITH APPLICABLE CMM.

TSN: UNK TSO: 0

Certifies that the work specified in Block 11/12 was carried out in accordance with EASA Part-145 and in respect to that work the component is considered ready for release to service under EASA Part-145 Approval Number: "EASA 145.6525".

13a. Certifies the items identified above were manufactured in conformity to:

Approved design data and are in a condition for safe operation.  
 Non-approved design data specified in Block 12.

14a.  14 CFR 43.9 Return to Service  Other regulation specified in Block 12

Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.

13b. Authorized Signature:

13c. Approval/Authorization No.:

13d. Name (Typed or Printed):

13e. Date (dd/mmm/yyyy):

14b. Authorized Signature:

14c. Approval/Certificate No.:

YNBR921L

14d. Name (Typed or Printed):

14e. Date (dd/mmm/yyyy):

SUSAN THOMPSON 2638442

06/Jun/2023

### User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

KEYSTONE TURBINE SERVICES  
CRS 8MHR895B  
WORK PERFORMED REPORT



KEYSTONE TURBINE SERVICES  
A **PAG** Company

885 FOX CHASE SUITE 111  
COATESVILLE PA. 19320  
1-610-268-6200  
1-484-786-8680 fax

Customer: MPAC Aviation Services  
Description: Fuel Control  
Serial #: C66799  
Work Order #: 85018901-EA-FC  
TSO:IN  
TSO:OUT  
Received date: 6/5/2023

Date: 7/14/2023  
Model #: DP-F2  
MFG Part #  
P/N #: 32444897-4  
P.O.#: 12077

Reason for removal: Overhaul

Run as Rec'd test: Not required

Warranty granted:	Approved	Warranty Review Concurrence
	Denied	
	N/A	T. Kline

Work Performed Overhauled & Tested IAW Honeywell CMM 73-20-78 Rev.2 6/8/21

Parts Replaced:

Part Number	Quantity	Description	Reason for replacement
KTS-FCU3244897-KI	1	Overhaul Kit	Req @ Overhaul

For KTS.

Date: 7/14/2023



**Action Aero**  
PO Box 22105  
Charlottetown, PE C1A 9J2  
Canada  
Ph: (902) 370-3311, Fax: (902) 370-3313  
info@actionaero.com

## Tear Down/Inspection

Work Order #: W119291

Date Printed: 5/30/2023

Time: 1:58:04 PM

Page: 1

Cust PO: 012073	PN: 3019906/25536-4	Descr: FLOW DIVIDER & DUMP VALVE	
Alt Cust PO	Serial #: 9959512673	Mfg: TRIUMPH	
Qty: 1	TSO UNK	TSNUNK	TSR UNK

### Symptoms:

Reason Removed

Flow divider removed for overhaul

Receiving Inspection

Flow divider received with fittings loose

### Faults:

Inspection Findings

No abnormal findings

### Corrective Actions:

Work Performed

Overhauled IAW CMM



**Action Aero**  
PO Box 22105  
Charlottetown, PE C1A 9J2  
Canada  
Ph: (902) 370-3311, Fax: (902) 370-3313  
info@actionaero.com

## Tear Down/Inspection

Work Order #: W119285

Date Printed: 6/21/2023

Time: 12:31:28 PM

Page: 1

Cust PO: 012074	PN: 702801-5	Descr: MAIN ENGINE FUEL PUMP
Alt Cust PO	Serial #: 001251	Mfg: ARGO-TECH
Qty: 1		
ESN: PCE-PC1288	TSO UNK	TSNUNK
		TSR UNK

### Symptoms:

Reason Removed

Fuel pump removed for overhaul

Receiving Inspection

Fuel pump received with a shipping cover, and with fittings attached

### Faults:

Inspection Findings

No abnormal findings

### Corrective Actions:

Work Recommended

Overhaul in accordance to the CMM

Work Performed

Overhauled unit IAW CMM.

PN	Description	Qty Needed	CD	Disposition	Status/Reason
93030-87	SEAL, STATIONARY	2	RPR	Issue	100%
AN6235-3A	FILTER ELEMENT, JT15D & PT6 FUEL PI	1	NE	Issue	100%



**Action Aero**  
PO Box 22105  
Charlottetown, PE C1A 9J2  
Canada  
Ph: (902) 370-3311, Fax: (902) 370-3313  
info@actionaero.com

## Tear Down/Inspection

Work Order #: W119287

Date Printed: 5/31/2023

Time: 10:03:17 AM

Page: 1

Cust PO: 012075

PN: 8210-002-01

Descr: TURBOPROP GOVERNOR ASSEMBLY

Alt Cust PO

Serial #: 13039147

Mfg: WOODWARD GOVERNOR

Qty: 1

ESN: PCE-PC1288

TSO UNK

TSNUNK

TSR UNK

### Symptoms:

Reason Removed

Governor removed for overhaul

Receiving Inspection

Governor received with a shipping cover

### Faults:

Inspection Findings

No abnormal findings.

### Corrective Actions:

Work Recommended

Overhaul IAW CMM with the incorporation of SB 33531C.

Work Performed

Overhauled IAW CMM with the incorporation of SB 33531C.

## INSPECTION REPORT

Page : 1 of 1

Close		<b>W/O No.</b>	
Abnormal/PH		9209881	
Normal/Risk			
Technician			
Inspector			
Aircraft Number		<b>P/N</b>	
A/C Position		10552E	
TSN	UNK	<b>S/N</b>	
TSO	UNK	WA23408	
TSR			
For Customer	PRIME TURBINES	<b>W/O Kind :</b>	[OVH] OVERHAULED
S.O. No.	9208241	Line No. :	1
Cust P.O.	MRO4242	P/N :	10552E
Our Ref	JACOB GARRATT	Desc :	FUEL HEATER PT6A
Stock		S/N:	WA23408
Shop	PAC Commercial	MFG :	STEWART WARNER SOUTH WIND CORP
Warr Claim		Qty :	1 Cond As: OH
Completed P/N	10552E	S/N As:	WA23408
Company	: 1	Department :	
Order Date	May-24-2023	Received Date :	May-18-2023 Due Date :
			Print Date : Jun-06-2023

### Overhaul WORK ORDER No. 9209881

Manuals	Description	Manual Date	Revision No.	Revision Date	Current	Until
73-10-03	FUEL HEATER	Aug-01-1975	16	Dec-15-2016	<input checked="" type="checkbox"/>	Feb-29-2024

Reported By Customer

CUSTOMER STATES OVERHAUL. RETURN SCRAP WITH UNIT.  
Preliminary Insp.

NO COMPONENT CARD.

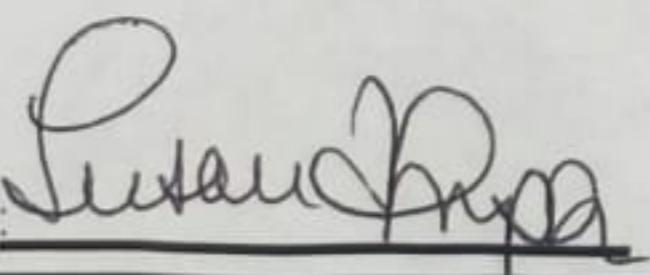
SENT UNIT FOR PAINT.  
Corrective Action

ASSEMBLED AND INSTALLED ALL CONSUMABLES.

PAINTED UNIT.

TESTED WITHIN LIMITS IN ACCORDANCE WITH APPLICABLE CMM.

Authorized Signature :



Name : Susan Thompson Date : 6 June 23

Updated:	08/01/2023	Name:	Brian Langeheine
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## Work Order # M5O1090

PCW	N/A	C/W	AD No:	Date	Description
	PT6A-114A				
	X		2022-08-13	05/27/2022	Turbine Section (CT Vane)
	X		2014-17-08R1	6/5/2015	Compressor Turbine CT Blades
X			2014-11-05	8/5/2014	Containment Ring Modification
	X		2012-09-10	5/23/2012	First Stage Sun Gears
X			2002-23-13	12/31/2002	Turbine Exhaust Ducts
	X		2001-20-01	11/5/2001	Compressor Bleed Valve Assembly
	X		97-04-12	3/14/1997	Compressor Bleed-off Valve
INSPECTOR: Brian Langeheine				DATE: 03/AUG/2023	

NOTE: This listing of FAA Airworthiness Directives may not always be up to date. Please refer to the [FAA Web Site](#) for the latest Airworthiness Directive information.

Abbreviations:		N/A	Not Applicable (model or part number)
PCW	Previously Complied With (found embodied)	C/W	Complied With (this shop visit)

## PT6 Service Bulletin Log Sheet

**Prime Turbines, LLC**  
FAA CRS No. 5TPR585C

<b>WORK ORDER</b>	M5O1090	<b>MODEL</b>	PT6A-114A
<b>SERIAL NUMBER</b>	PCE-PC1288	<b>BUILD SPEC</b>	750

**S/Bs FOUND PREVIOUSLY INCORPORATED:**

## S/B'S INCORPORATED THIS VISIT:

I HEREBY CERTIFY THAT THE ABOVE LISTED SERVICE BULLETINS HAVE EITHER BEEN FOUND PREVIOUSLY INCORPORATED BASED ON RECORDS RESEARCH OR WERE INCORPORATED ON THIS WORK ORDER IN ACCORDANCE WITH FEDERAL REGULATIONS.

<b>Inspector:</b>	Brian Langeheine	<b>Date:</b>	03/AUG/2023
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