



PT. SMART CAKRAWALA AVIATION
WORK ORDER
Form: SCA/MTC/030

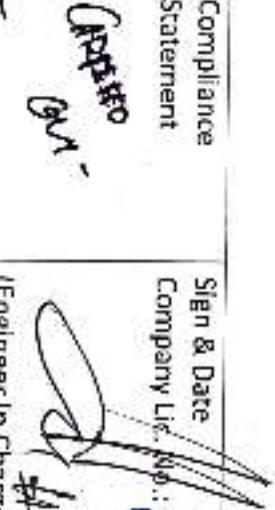
Subject :	No.	WO/058-SNV/X/2023
Engine Replacement	Date	12 Oct 2023
Reference :	A/C Reg.	PK-SNV C208B-5551
MP C208B Issued 01	Prepared By	TS
	Checked By	CI
To : Engineer In Charge	Approved By	TM

Description:

1. Perform Engine Replacement
2. Make an entry in Maintenance Log.
3. Return the Completed Work Order and Form to PPC.

#If any finding, please close the routine card, and transferred to Inspection card.

Additional Work:

Compliance Statement	Sign & Date Company Ltd. No.:  	Signature  (Technical Manager)
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AIRCRAFT CHECK WORK SUMMARY
(Form: SCA/MTC/051)

DATE OF ISSUED	JOMO #	TYPE OF MAINTENANCE	DATE OF ACCOMPLISHED	
12 Oct 2023	WO/058-SNW/X/2023	Replacement	19 Oct 2023	
AC Type C208B		Mfg. Serial Number C208B-5551	AC Registration PK-SNW	
AIRCRAFT DATA				
Subject	Pos #	Serial Number (SN)	TTSN/TCSN	
Engine	#1	PCE-MA0607	2553 : 07 / 9186.	
Propeller/Rotor	#2	-	4999 : 00 595 : 29.	
Landing Gear	LH MLG	NLG	4999 : 00 / 6700	
	RH MLG		4999 : 00 / 690	
PACKAGE COVERED				
No	Subject	Gty	Remark	
1	Non Routine Card	1		
2	Inspection Card	1		
3	Work Order	1		
4	Summary Inspection List	1		
5	Material and Tool List	-		
6	Escalation form	-		
7	CRS (SMI 'Unscheduled Maintenance)	1		
INSPECTION CARD (IC) LIST (Finding during maintenance)				
No	Taskcard Ref	Subject	Status	Name/ Sign & Stamp
IC-001			Open	
IC-002			Close	
IC-003				
IC-004				
IC-005				
IC-006				

<u>IC-007</u>				
<u>IC-008</u>				
<u>IC-009</u>				
<u>IC-010</u>				
<u>IC-011</u>				
<u>IC-012</u>				
<u>IC-013</u>				
<u>IC-014</u>				
<u>IC-015</u>				

Prepared by:
Technical Support

Checked by:
Chief Maintenance

Verified by:
Chief Inspector

Approved by:
Technical Manager

Hari

Dodil

Yanuar

Istiono



SUMMARY INSPECTION ITEMS
(Form: SCA/MTC/050)

WO Ref: WO/058-5NU/X/2023

NO.	TASK CARD NO.	DESCRIPTION	DATE	EST MHR	NAME	STAMP
1	NRC-01	ENGINE ASSY REPLACEMENT REMOVED PN: PT6A-140 SN: PCE-V40807 INSTALLED PN: PT6A-140 SN: PCE-V40567	19/10/23	10	Patricia	25
2	APPENDIX B05	COMPASS SWING	19/10/23	10	Patricia	27

MAINTENANCE PROGRAM

CESSNA 208/208B



Appendix B05 – Magnetic Compass Calibration

Reg. Mark	PK - GAW	Date	6710
MSN	C408B - 5551	Station	NADIR
TSN / CSN	A999 : 10	WO No.	WO/058- SWV/X/2023

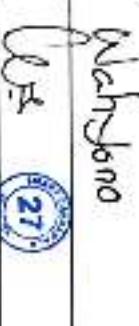
NO.	TASK	SIGNATURE	
		ENGINEER	RII SIGN&STAMP
01	Magnetic Compass Functional Check. Refer to AMM 34-21-00.		
02	Record the Magnetic Compass functional check result, calculate and make an entry in form SCA/MTC/026.		

PERSONNEL PARTICIPATING IN THIS INSPECTION

NAME	POSITION	SIGNATURE	LICENSE NUMBER
Denny Pramono	ENGINER		2077

RETURN TO SERVICE

The work recorded above has been carried out in accordance with the requirements of the Civil Aviation Safety Regulation for the time being in force and in that respect the aircraft is consider fit for Release to Service.

Name : Wahyono
Place/Date : Nabire, 19-10-2023.
Sign & Stamp : 




NON ROUTINE CARD
[Form: SCA/MTC/047]

1. JDN/DOY	2. DATE	3. MTC TYPE	4. AC REG/NSN
W0059-SNVX/2023	19 Oct 2023	REPLACEMENT	PK-SNV/2008-5551
5. CARD #	6. ATA SPEC	7. TRADE	8. STA
#001	80		
9. ZONE	10. PANEL		
ENGINE			
<p>11. DESCRIPTION</p> <p>ENGINE ASSY REPLACEMENT</p> <p>REMOVED PIN: PT6A-140 SN: PCE-VA0607</p> <p>INSTALLED PIN: PT6A-140 SN: PCE-VA0567</p>			
REFERENCE	<input checked="" type="checkbox"/> EMM Ch. 72-00-40	<input type="checkbox"/> AMM Ch.	<input type="checkbox"/> OTHER
R/N (*)	<input checked="" type="checkbox"/> ✓	<input type="checkbox"/> N	MHR:
<p>12. RESULT</p> <p>DISASSEMBLED ENGINE RE PLACE INSTR. FRM PT6A-140 Reg. 50 FRM C1088 Reg. 59</p> <p>Performed at ACT TT: 41.99 :10 ACT ALDG: A910</p>			
FINDING	<input type="checkbox"/> ✓	<input type="checkbox"/> N	ACT MHR:
INSPECTION CARD (IC) #		19	10 2023
<p>12. PARTS REQUIRED</p>			
DESCRIPTION	PART NO	QTY	REMARK
Ref 100 1058-5444 / X/2023	N/A	-	<input checked="" type="checkbox"/> ✓
<p>14. TOOLS REQUIRED</p>			
DESCRIPTION	PART NO / MODEL	NEXT CALIBRATION DATE	STATUS



MAINTENANCE PROGRAM

CESSNA C208/C208B

Appendix B03 – PT6A-140 Engine Run Performance Sheet

Reg. Mark : PK-SNU

WO/FM/No. : W0658-SNU/28/2023

PRE - INSPECTION		POST - INSPECTION	
Location	NAFIAK	NAFIAK	NAFIAK
Date	14 Oct 2023	19 Oct 2023	19 Oct 2023
Cycle			
Filed Barometric	1010	1009	1009
OAT	32	23	23
Altitude	20 ft	30 ft	30 ft

PRE - INSPECTION		POST - INSPECTION	
	Target		Actual
Tq	1347	1219	2397
Np	1960	1840	1960
IT	84 °C	83.5 °C	83.9 °C
Ng	101.7 %	99.6 %	102.7 %
Wf	540	510	548
Oil Press	20	15	21
Oil Temp	71 °C	63 °C	63 °C
Start Temp	72.5 °C	70.4 °C	70.4 °C

Engine Run Up Checks
Inertial <input checked="" type="checkbox"/>
EPL <input checked="" type="checkbox"/>
OVG <input checked="" type="checkbox"/>
Stby Alt <input checked="" type="checkbox"/>
BOV <input checked="" type="checkbox"/>
Brake <input checked="" type="checkbox"/>
Randown <input checked="" type="checkbox"/>

NOTE:

1. Brake system at Torque 2000 ft-lbs. 3. CPL check can't exceed 48.9% air forward. 5. Low abs at 55.5 - 57% air forward. 6. High abs at 64 - 66% air forward.

2. Inertial Separator at Torque 2000 ft-lbs.

Engine Performance Target Table (Cessna C208B EX)

OAT (°C)	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41
Tq (ft-lbs)	2357	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397	2397
Np	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900	1900
IT (°C)	83.5	83.7	83.9	84.1	84.1	84.1	84.1	84.1	84.2	84.3	84.3	84.5	84.6	84.6	84.6
Ng (%)	102.7	102.7	102.7	102.7	102.7	102.5	102.6	102.6	102.6	102.5	102.5	102.5	102.5	102.5	102.5
Wf (lb)	578	578	578	578	578	578	578	578	578	578	578	578	578	578	578

Note:

1. Make sure that inertial separator in normal condition, no bleed air extracted from the engine and air condition OFF.
2. This table only applies to altitude 0-500 feet MSL. For higher altitude, refer to EMM 72-00-00.
3. Max fuel flow is 580 lb/hr fuel flow is not more than 15 lbs/hr higher than the value shown in table.
4. If parameters are outside the target, performance table to EMM chapter 71-00-10.

REMARKS	LOW	HIGH	DATE
Hi sp. idle	57.0	64.8	10/2023
Ref. 100% IT	22.06	27.1	10/2023
Ref. 100% Wf	578	578	10/2023

PERFORMED BY			
Name	Sig. & Stamp	Date	Location
	19/10/2023	19/10/2023	NAFIAK



Aircraft Registration: **PK-SNV**

PK-SNV

WO# Nr: WO/058-SNV/X/2023

Additional Work Sheet
Engine Replacement

Parts Used Sheet

Special Tool Used

No.	Description	Special Tool Used	Due Date	Engineer
1	TORQUE WRENCH.	P/N 10050100004	NOV 2023	Devin
2	TORQUE WRENCH.	P/N 10050100004	NOV 2023	Devin
3	TORQUE WRENCH	P/N 10050100004	SEPT	Devin
4.	OK PULLEY.		O/C	Devin
5	PROPELLER PULLEY.		O/C	Devin
6	ENGINE SLING		O/C	Devin
7.	COMPASS SWING P/N 1686 - REV S/N 751823/05		NOVEMBER 2023	Devin



Aircraft Registration:

PK-SNV

Additional Work Sheet

Engine Replacement

Parts Used Sheet

WO# Nr: WO/058-SNV/X/2023

Date	Part Nr.	Serial Nr.	Description	Quantity	Engineer
14-OCT-2023	ENGINE REPLACEMENT KIT		ENGINE REPLACEMENT KIT	1 EA	DENIAK
15-OCT-2023	EASTMAN TURBO OIL		EASTMAN TURBO OIL	12 QUART	NINDO
14-OCT-2023	M83248/2-904		O-RING	1 EA	DENIAK
14-OCT-2023	A1633-72		O-RING PROP	1 EA	NINDO
14-OCT-2023	MS28775-022		O-RING	1 EA	DENIAK
14-OCT-2023	MS29665-132		COTTER PIN	1 EA	NINDO
14-OCT-2023	MIL W-5-632		PENSTILUBE	1 EA	DENIAK
14-OCT-2023	A1633-72		NUT PROP	2 EA	NINDO



PT. SMART CAKRAWALA AVIATION

CERTIFICATE RETURN TO SERVICE
SCHEDULED MAINTENANCE INSPECTION
(FCC GRM)

SCHEDULED MAINTENANCE INSPECTION

**CERTIFICATE RETURN TO SERVICE
SCHEDULED MAINTENANCE INSPECTION
(CRS-SMI)**

A/C TYPE	CESSNA 208B	TSN	449910
A/C REG	PK-SNV	TCN	6710
MSN	C208B-5551	DATE	19-08-2013

TYPE OF INSPECTION	ENGINE REPLACEMENT
DUE AT	4000 HOURS ON STE 4200 HOURS
REF	MP 208/2008 ISSUED 01
EXCEPTION	

AUTHORIZED PERSON
is been maintained accordance with
Aircraft safe and airworthy for flight

Aircraft sale and airworthy for flight				
NAME	CAT	AMELIOTR NO	SIGN & STAMP	DATE
Dewey - Powers	AIRFRAME & POWER PLANT	1607/026	 	10/21/1923
	EIRA			

THE NEXT DUE TYPE OF INSPECTION : EXHALE DEPLACEMENT

DUE AT

Form: SCA/SMTC/049



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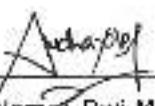
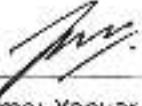
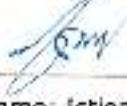
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ENGINEERING ORDER

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**REMOVAL & INSTALLATION OF ENGINE
ASSY PT6A-140 ON CESSNA 208B**

PT. SMART CAKRAWALA AVIATION

Prepared	Checked	Approved
Technical Support	Chief Inspector	Technical Manager
Signature: 	Signature: 	Signature: 
Name: Dwi M.	Name: Yanuar A. Fatah	Name: Istiono
Date: 12 Oct 2023	Date: 12 Oct 2023	Date: 12 Oct 2023



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**SMART AVIATION
ENGINEERING ORDER**

No. EI:
015/EO/TEK-TS/X/2023

Rev. No. :
ORIGINAL

Date Issued :

October 12, 2023

Task Description :

**INSTALLATION OF ENGINE ASSY PT6A-140
ON CESSNA 208B**

Data Reference :

- **Model 208 Series Maintenance Manual
Revision 39, Revision Date Mar 1, 2023
Chapter 71 Power Plant – Maintenance
Practices**

Aircraft Type :

**CESSNA 208B WITH ENGINE MODEL PT6A-
140**



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**SMART AVIATION
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1. Description.

This EO is issued, to perform removal and installation checklist powerplant maintenance practices the PT6A-140 engine on Cessna 208B.

2. Aircraft Effectivity.

REGISTRATION	SERIAL NUMBER
PK-SNV	208B-5551

3. Compliance

The Engine model PT6A-140 have TBO 4000 Hours, do removal and installation the overhauled/new engine on the aircraft refer to accomplishment instruction task card.

4. Distribution.

TECHNICAL MANAGER	[<input checked="" type="checkbox"/>]	MATERIAL SUPPORT	[<input checked="" type="checkbox"/>]
SAFETY & QUALITY MANAGER	[<input checked="" type="checkbox"/>]	TECHNICAL SUPPORT	[<input checked="" type="checkbox"/>]
CHIEF INSPECTOR	[<input checked="" type="checkbox"/>]	FILE	[<input checked="" type="checkbox"/>]

5. Manhours

32.0 man-hour to do the inspection.



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6. Material.

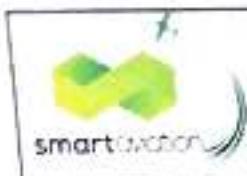
3074153-01 PROPELLER GOVERNOR
A 1633-72 O-RING HUB TO PROPELLER SHAFT
A 1639-32 NUT
B 5096 SPACER
B 5121 FEEDBACK ASSY
MS 206685 GASKET PROPELLER OVER SPEED GOVERNOR
206684G or 303952 GASKET PROPELLER GOVERNOR
AN 4044-1 GASKET STAR-GEN
S 3346-1 GASKET PROPELLER TACHOMETER
S 3346-1 GASKET NG TACHOMETER
S 3346-1 GASKET STBY ALTERNATOR
S 3346-1 GASKET AC COMPRESSOR DRIVE UNIT
AN363-720 NUT
MS24665-302 COTTER PIN MOUNT BRACKET TO MOUNT RING
VSF1015N12B SEAL CONICAL
9910333-1 ELASTOMER
MS24665-302 COTTER PIN
MS24665-134 COTTER PIN
MS24665-86 COTTER PIN
3007342 GASKET
S2808/AE3663 HDSE OIL
MIL PRF 83483C LUBRICANT FOR THREAD PROPELLER
MIL W-G-632 LUBRICANT FOR COMPRESSOR DRIVE UNIT, PLASTILUBE
2380 ENGINE OIL

7. Special Tool Required.

Propeller Special tool D-5945 1 SET
7/8 inch special tool 1 SET
MASTER COMPASS

8. Publication Affected.

None.



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**SMART AVIATION
ENGINEERING ORDER**

9. Accomplishment Instructions.

C208B ENGINE REMOVAL

Date : 14-07-2023 Work Number : 100/058-SNV/X/2023
Part No. Engine : PT6A-140 A/C Total Hours : 4193:10
Ser. No. Engine : PCE-VA0607 A/C Total Landings : 67/0
Engine Time TSN: TSO:N/A
CSN: CSO:N/A

Removed from A/C Reg. : PK-SNV

Description	Eng.	RII	Remarks
A. REMOVE ENGINE (Refer to Figure 201 and Figure 202)			
CAUTION: Check main wheels and place a tail stand under tailcone before attempting engine removal.			
1. Remove external electrical power from the airplane.			
2. Pull fuel firewall shutoff control out (off).			
3. Remove the cowling components as follows. Refer to Chapter 71, Engine Cowling and Nose Cap - Maintenance Practices: (a) The upper cowling doors. (b) The lower cowling panels. (c) The right nose cap.			
4. Use the filter drain to drain the residual fuel from lines and fuel filter. Refer to Chapter 28, Fuel Lines, Valves and Filters - Maintenance Practices.			
5. Remove the fuel supply hose at the fuel heater. Refer to Chapter 73, Oil-to-Fuel Heater - Maintenance Practice section of the Pratt and Whitney PT6A-140 Maintenance Manual P/N 3075742 found in the Introduction List of Publications.			
6. Remove the fuel motive flow hose at the motive flow shut off valve. Refer to Chapter 73, Fuel Control Unit - Maintenance Practices section of the Pratt and Whitney PT6A-140 Maintenance Manual P/N 3075742 found in the Introduction List of Publications.			
7. Remove the oil cooler. Refer to Chapter 79, Oil Distribution - Maintenance Practices (PT6A-140).			



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8. Remove top cowl center panel assembly and nose cap. Refer to Chapter 71, Engine Cowling and Nose Cap - Maintenance Practices

[Handwritten signature]

9. Remove the propeller. Refer to Chapter 61, Propeller (Hartzell) Maintenance Practices.

[Handwritten signature]

10. Disconnect and remove propeller speed control cable. Refer to Chapter 76, Quadrant Assembly And Controls - Maintenance Practices.

[Handwritten signature]

11. Remove the left nose cap/induction air duct/inertial air separator. Refer to Chapter 71, Inertial Air Separator - Maintenance Practice

[Handwritten signature]

13. Disconnect the cabin heater bleed air line at the flow control valve. Chapter 21, Compressor Bleed Air Heater - Maintenance Practices.

[Handwritten signature]

14. Disconnect the bleed air hose at mixing air valve. Chapter 21, Compressor Bleed Air Heater - Maintenance Practices

[Handwritten signature]

15. Remove the starter/generator cooling air hose from starter/generator. For the 300 Amp Starter/Generator refer to Chapter 80, 300 AMP Starter/Generator Cooling Air Duct - Maintenance Practices. For the 200 Amp Starter/Generator refer to Chapter 80, Starter/Generator Cooling Air Blast tube - Maintenance Practices

[Handwritten signature]

16. Remove the oil pressure switch supply hose. Refer to Chapter 79, Oil Pressure Switch - Maintenance Practices

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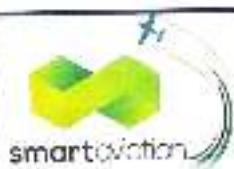
17. Remove engine fire detector wiring harness. Chapter 26, Fire Detection System - Maintenance Practices

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18. Disconnect electrical wiring connectors and ground wires at the following equipment locations:

- Battery Connector (PN004) (aft right side of engine)
- Prop Overspeed Valve Connector (PN041) (left front of engine)
- NP Speed Tach (PN033) (right front of engine)
- Cabin Heat Bleed Air Valve connector (PN043) (lower right side of engine)
- Oil Pressure Switch (PN030) (right side on engine truss)
- Oil Temperature Sensor connector (PN031) (right rear of engine)
- NG Speed Tach (PN034) rear, (lower right side of engine)
- Starter Generator Connector (PN002) (center top of engine accessory case)
- Ignition Exciter Connector (PN040) (right engine mount truss)
- Fuel Flow Connector (PN032) (rear, lower right side of engine)
- Torque Transducer (PN038) (right engine mount truss)

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- RGB Chip Detector (PN035) (right engine mount truss)
- Engine ground straps airplane frame connections.

19. Disconnect the engine power control cables at fuel control unit. Refer to Chapter 73, Fuel Control Unit - Maintenance Practices section of the Pratt and Whitney PT6A-140 Maintenance Manual P/N 3075742 found in the Introduction List of Publications.

20. Remove torque meter pressure and vent lines at forward upper right side of engine mount truss. Refer to Chapter 77, Wet Torque Indicating System - Maintenance Practices (PTGA-140).

21. Connect hoist sling to forward and aft lifting brackets and connect sling to engine hoist.

22. Raise hoist to just support weight of engine and remove nuts and bolts at each of four corners of engine mounting ring.

23. Make sure that all wiring and lines are free, then carefully move hoist and engine forward to clear engine mount truss.

24. If engine is to be returned for overhaul or replaced refer to Prepare Engine to Send for Service.

***** END OF THE TASK *****



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**SMART AVIATION
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C208B ENGINE INSTALLATION

Date : 14-OCT-2023 Work Number : W0/058-5KN/X/2023
 Part No. Engine : PT6A-140 A/C Total Hours : 4199:10
 Ser. No. Engine : PCE-VA0567 A/C Total Landings : 6710
 Engine Time TSN: 2553:70 H TSO: N/A
 CSN:9386 CSO: N/A
 Installed on A/C Reg. : PK-SNV

Description	Eng.	RII	Remarks
-------------	------	-----	---------

25. INSTALL ENGINE (Refer to Figure 201 and Figure 202).

1. If the engine is new or was at manufacturer for service, install engine components. Refer to Engine Build-Up.
2. Make and inventory record P/N and S/N of the engine and its accessories. Fill out into the List (Form: Engine Change – Major Component Inventory Record)
3. Install engine mount brackets, elastomers, and engine mount ring. Refer to Chapter 71, Engine mount – Maintenance Practices.
4. Connect the lifting hoist sling to forward and aft lifting brackets on engine and lift engine in its correct position forward of engine mount.
5. Make sure that all engine lines and equipment are clear.
6. Lubricate the engine mount bolts with M L-PRF-81322G Grease, before you install them to prevent corrosion.
7. Make sure that the threads of bolts are covered during application of grease. Lubrication on threads can alter the torque reading.
8. Move the hoist and engine aft to align the engine mount ring holes with the holes in the engine mount truss.
9. Install the mount bolts and torque the bolt/nuts to 480 to 690 inch-pounds.
 - a. Remove the hoist and sling
10. Connect torque meter pressure and vent lines at upper left Firewall
 - a. Do a leak test of the torque meter pressure and vent lines. Refer to Chapter 77, Wet Torque indicating System - Maintenance Practices



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11. Connect engine power controls at fuel control unit.
(a) Rig the fuel controls. Refer to Chapter 76, PT6A-140 Engine Power Control Rigging - Adjustment/Test.

12. Connect the electrical wiring connectors and ground wires at the following equipment locations:

- Battery Connector (PN004) (aft right side of engine)
- Prop Overspeed Valve Connector (PN041) (left front of engine)
- NP Speed Tach (PN033) (right front of engine)
- Cabin Heat Bleed Air Valve connector (PN043) (lower right side of engine)
- Oil Pressure Switch (PN030) (right side on engine truss)
- Oil Temperature Sensor connector (PN031) (right rear of engine)
- NG Speed Tach (PN034) rear, (lower right side of engine)
- Starter Generator connector (PN002) (center top of engine accessory case)
- Ignition Exciter Connector (PN040) (right engine mount truss)
- Fuel Flow connector (PN032) (rear, lower right side of engine)
- Torque Transducer (PN038) (right engine mount truss)
- RGB Chip Detector (PN035) (right engine mount truss)
- Engine groundstraps airplane frame connections.

13. Install the oil pressure switch supply hose. Refer to Chapter 79, Oil Pressure Switch - Maintenance Practices

14. install the engine fire detector wiring harness. Chapter 26, Fire Detection System - Maintenance Practices

15. Install the starter/generator cooling air hose to the starter/generator. For the 300 Amp Starter/Generator refer to Chapter 80, 300 AMP Starter/Generator Cooling Air Duct - Maintenance Practices. For the 200 Amp Starter/Generator refer to Chapter 80, Starter/Generator Cooling Air Blast Tube - Maintenance Practices.

16. Connect the cabin heater bleed air line at the flow control valve. Chapter 21, Compressor Bleed Air Heater - Maintenance Practices.

17. Connect the bleed air hose at mixing air valve. Chapter 21, Compressor Bleed Air Heater - Maintenance Practices

18. If necessary, install the left nose cap/induction air duct/inertial air separator. Refer to Chapter 71, Inertial Air Separator - Maintenance Practices

19. If necessary, install the propeller. Refer to Chapter 61, Propeller (Hartzell) - Maintenance Practices.



TECHNICAL SUPPORT
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**SMART AVIATION
ENGINEERING ORDER**

20. Install the propeller speed control cable. Refer to Chapter 76, Quadrant Assembly And Controls - Maintenance Practices.

[Signature]

[Signature]

21. Install left and right nosescap bulkhead assemblies and top cowling center panel. Refer to Chapter 71, Engine Cowling and Nose Cap - Maintenance Practices.

[Signature]

[Signature]

22. Install the oil cooler. Refer to Chapter 79, Oil Distribution - Maintenance Practices (PT6A-140).

[Signature]

[Signature]

23. Install the right nosescap. Refer to Chapter 71, Engine Cowling and Nose Cap - Maintenance Practices.

[Signature]

[Signature]

24. Install the fuel supply hose at the fuel heater. Refer to Chapter 73, Oil-To-Fuel Heater Unit - Maintenance Practices section of the Pratt and Whitney PT6A-140 Maintenance Manual P/N 3075742 found in the Introduction List of Publications.

[Signature]

[Signature]

25. Install the fuel motive flow hose at the motive flow shut off valve (MFSOV). Refer to Chapter 73, Fuel Control - Maintenance Practices section of the Pratt and Whitney PT6A-140 Maintenance Manual P/N 3075742 found in the Introduction List of Publications.

[Signature]

[Signature]

26. Purge the fuel lines as follows:

- Push fuel firewall shutoff control fully in.
- Disconnect the supply fuel line at fuel manifold below engine. 1 Cap down stream line. 2 Use correct shop practices to collect fuel from open supply line.
- Use the starter to motor the engine.
- When the purge is complete, connect the fuel line to the manifold.

Use correct shop practices to discard purged fuel.

[Signature]

27. Do an operational check of the different components on the engine.

- Start the engine and do the operational check. Refer to Pilot's Operating Handbook and FAA-Approved Airplane Flight Manual and the Pratt and Whitney PT6A-140 Maintenance Manual P/N 3075742 found in the Introduction List of Publications.

[Signature]

- Complete an Engine Performance Check. Refer to Chapter 71-00-05, Power Plant (PT6A-140) - Adjustment/Test, Engine Performance Check.

28. Perform RI inspection if any controls have been disturbed or adjusted.

[Signature]

[Signature]

29. Install engine cowling.



TECHNICAL SUPPORT
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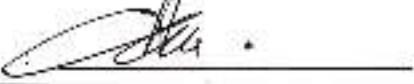
**SMART AVIATION
ENGINEERING ORDER**

30. Make an appropriate entry in Work Order (WO) and Aircraft Flight
Maintenance Log (AFML)

MAINTENANCE RELEASE

I hereby certify that the above stated maintenance and/or inspection was performed in accordance with
the approved Aircraft Maintenance Program and meets requirements of Civil Aviation Safety Regulations.

Name : Denny Rivera Stamp :

Signature :  Place/Date : NAC/AF 19 oct 2023



TECHNICAL SUPPORT
TECHNICAL DEPARTMENT
ENGINEERING ORDER

	015/EO/TEK- TS/X/2023
Rev. No	ORIGINAL
Rev. Date	12/10/2023

ENGINE CHANGE - Major Component Inventory Record

Registration	:	PK-SNU	Work Order Number	:	W6 / 085 - SNU / X / 2023
Airframe Time	:	490 : 0	Airframe Landings	:	670
Engine Time	:	4195 : 0	Engine Cycle	:	670

Description	Part Number	Engine OFF		Engine ON	
		Serial Number	Time Remaining	Part Number	Serial Number
Engine Assembly	PT6A-140	PCF-VAD607		PT6A-140	PCF-VAD607
Propeller Assembly	P778550-01	18192		P778550-01	18192
Compressor Bleed Valve	S40-1407-4	9460		3094038-03	83388
Fuel Control Unit	8063-636	22386669		8063-636	2159434
Oil Fuel Heater	103814	WA59506		103814	WA58660
Igniter Exciter	CA92106-14	190820		CA92106-34	190810
Flow Divider	25628-3	11114		25628-3	M0500
Oil Cooler	380395000	70558		380345000	70558
Starter Generator	300561450	500103		300561450	500103
Alternator	9910532-2	1-7031772		9910591-2	H-1031772
Fuel Pump	3090760	016863		3040760P	016863
Propeller Governor	66503.820-000-01	21681009		8110-002-01	15318152
Propeller Overspeed Governor	66503.210539-01	21669988		66503-210539-01	21551615
Fuel Nozzle	3077780-01A	311836A			

NOTE: ANY OTHER COMPONENT CHANGES MUST BE FILLED ON INSPECTION CARD (SCA/MTC/048)

1. Approving National Aviation Authority/Country: FAA/UNITED STATES	2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG				3. Form Tracking Number: 235344
4. Organization Name and Address: PRATT & WHITNEY ENGINE SERVICES, INC. 1525 MIDWAY PARK RD BRIDGEPORT, WV 26330 FAA CRSB1 MIR301K					5. Work Order/Contract Number: 235344
6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial/Batch Number:	11. Status/Work:
ONE	ENGINE, TURBOPROP PT6A-140	3076226-01	ONE	PCE-VA0567	TESTED
12. Remarks:	TTSN: 2553.7	TCSN: 9386	TTSO: N/A	TCSO: N/A	

This engine has tested as received in accordance with P&WC Overhaul Manual P/N 3075743, Rev. 50, dated 24 October 2022; and other P&WC approved documentation and/or engineering documentation acceptable to the administrator.

Engine test certificate and list of parts replaced are attached to the logbook. Pertinent details of the work performed are on file at this repair station under the above sales order number.

Certifies that the work specified in block 11/12 was carried out in accordance with EASA part 145 and with respect to that work the aircraft engine is considered ready for release to service under EASA acceptance certificate number: EASA.145.4757.

Export classification: ECCN 9E991

13a. Certifies the items identified above were made in accordance with: <input type="checkbox"/> Approved design data and airworthiness qualifying for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.	14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature:	14c. Approval/Certificate No.:
13d. Name (Typed or Printed):	13e. Date (dd/mm/yyyy):	14d. Name (Typed or Printed):	14e. Date (dd/mm/yyyy):

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

Pratt & Whitney Engine Services, Inc.

ENGINE FUEL SYSTEM PRESERVATION

ENGINE NO: PCE-VA0567

THIS FUEL SYSTEM HAS BEEN PRESERVED WITH OIL

MIL-L-6081

GRADE 1010

DEPRESERVE IN ACCORDANCE WITH THE APPLICABLE MAINTENANCE
MANUAL PROCEDURE

INSPECTOR: Scott Bublitz

PWV
75

DATE: 18 July 93

PWV 3044 (OCT 99)

1. Aspiring National Aviation Authority X entries:

FAA UNITED STATES

AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

4. Organization Name and Address: PRATT & WHITNEY ENGINE SERVICES, INC.
1525 MIDWAY PARK RD BILLINGSBELT, WY 82619
FAA CIRSP LMIR301K

5. Form Tracking Number:

235344

6. Work Order/Contract/Job
Number:

235344

7. Description:	8. Part Number:	9. Quantity:	10. Serial/Block Number:	11. Status/Work
ONE ENGINE, TURBOPROP PT6A-140	3076226-01	ONE	PCE-VAD567	TESTED
12. Remarks: TTSN: 2551.7 TCSN: 9386 TTSO: N/A TCSO: N/A				

This engine has tested as received in accordance with P&WC Overhaul Manual P/N 3075745, Rev. 50, dated 24 October 2022; and other P&WC approved documentation and/or engineering documentation acceptable to the administrator.

Engine test certificate and list of parts replaced are attached to the logbook. Pertinent details of the work performed are on file at this repair station under the above sales order number.

Certifies that the work specified in block 11/12 was carried out in accordance with EASA part 145 and with respect to that work the aircraft engine is considered ready for release to service under EASA acceptance certificate number: EASA 145.4757.

Export classification: ECCN 91991

13a. Certifies the items identified above were manufactured in accordance:

- Approved design drawings are in a condition for safe operation.
- Non-approved design data specified in Block 12.

14a. 14 CFR 43.9 Return to Service

Other regulation specified in Block 12

Certifies that unless otherwise specified in Block 13, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.

13b. Authorized Signature:

13c. Approved/Authorizations No.:

13d. Date (dd/mm/yy):

14b. Authorized Signature:

14e. Appendix certificate No.:

LMIR301K

14f. Name (dd/mm/yy):

14g. Date (dd/mm/yy):

24 Jul 2023

14d. Name (Typed or Printed):

14e. Name (Typed or Printed):

Scott Bucklew, Authorized Inspector

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically confer authority to install the part/component/assembly.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority approves parts/component/assembly from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 1a and 1b do not constitute installation certification. In all cases, aircraft maintenance records and certain installation certificates issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

Pratt & Whitney Canada

Canada



liste des séminaires motivant à nombreux de série

www.ijerph.org

Engine Serialized Component Summary

Pratt & Whitney Canada  www.pwcanada.com 1-800-666-9600

1

Model	P76A-140	N° de série	Serial No.	Build Spec No.	PC-E-VAD567	Caract. de matériau	N° de série du matériau	Designation du matériau	N° du matériau	Matériau	N° de lot pour l'achat	N° de fournisseur	N° du matériau du fournisseur	N° du matériau
3011095	SHFT-POWER TURBINE	RWA60276	PXAHX	EPWPE	M1139	HRAG74346	ERAEPPWE40	TEMPERATURE SENSOR	DISC-COMPRESSOR	DISC-COMPRESSOR	390	40		
3027629	FLOW DIVIDER AND DUMP VALVE	RWA60276	PXAHX	EPWPE	M1139	HRAG74346	ERAEPPWE40	IMPELLER CENTRIFUGAL	DISC-COMPRESSOR	DISC-COMPRESSOR	390	390	3027796	3037603
3029670	BUS BAR-THERMOCOUPLE ASSEMBLY	RWA60276	PXAHX	EPWPE	M1139	HRAG74346	ERAEPPWE40	TEMPERATURE SENSOR	DISC-COMPRESSOR	DISC-COMPRESSOR	390	390	3040213	3040312
3040760	PUMP FUEL	Q1NS85	Q1NS85	EPWPE	182626	9326	HLA447AC	VANE RING-POWER TURBINE ASSY	SHAFT-STUB,COMPRESSOR ROTOR	SHAFT-STUB,COMPRESSOR ROTOR	3055649-01	3055649-02	304937-03	304937-06
304937-03	VALVE-COMPRESSOR BLEED	9326	182626	EPWPE	NR00718	NR00718	HLA447AC	VANE RING-POWER TURBINE ASSY	SHAFT-STUB,COMPRESSOR ROTOR	SHAFT-STUB,COMPRESSOR ROTOR	3055649-01	3055649-02	304937-06	304937-09
304937-06	EXCITER-IGNITION	EPWPE	EPWPE	EPWPE	FE00361	HLA2318B707	VANE RING-POWER TURBINE ASSY	TORQUE LIMITER	GEARING-CHLFLG 137484X2 672X 5906	GEARING-CHLFLG 137484X2 672X 5906	3073900-01	3073900-02	3073900-03	3073900-04
3073900-01	BLADE-POWER TURBINE	YWWAR37	YWWAR37	EPWPE	AE80299533	AE80299533	AE80300253	BEARING-SBALL,FLG. 1810X2830X2 745	GOVERNOR-PROPELLER	GOVERNOR-PROPELLER	3074154-01	3074154-02	3073900-03	3073900-04
3074154-01	SLADE-POWER TURBINE	YWWB73	YWWB73	EPWPE	YWWB6	YWWB6	YWWB24	YWWB24	YWWB25	YWWB25	YWWB26	YWWB26	YWWB27	YWWB27
3074154-02	SLADE-POWER TURBINE	YWWB73	YWWB73	EPWPE	YWWB6	YWWB6	YWWB24	YWWB24	YWWB25	YWWB25	YWWB26	YWWB26	YWWB27	YWWB27

Liste des composants moteur à numéros de série Engine Serialized Component Summary



Pratt & Whitney Canada

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Modèle Model	PT6A-140	N° de série Serial No.	PCE-VA0567	Chapt. de montage Build Spec No	BS1294	
N° du matériel Material Number	Désignation du matériel Material Name	N° de série Serial No.	Code de trait. thermique Heat Code	N° de série du matériel forgé forging Serial No.	N° de lot pour trait. thermique Heat Treat Batch No.	N° de matériel du fournisseur Vendor Material No.
3011095	SHAFT-POWER TURBINE	RWAGDF278	PXAHX			
3022670	FLOW DIVIDER AND DUMP VALVE	M1139				
3027629	BUS BAR-THERMOCOUPLE-ASSY	HRAG74346				
3027798	IMPELLER-CENTRIFUGAL	FRAEPMP40	EPWPE	40		
3037603	TEMPERATURE SENSOR	NR06304				
3040213	DISC-COMPRESSOR	ADACBBC				
3040312	DISC-COMPRESSOR	ADACBBC				
3040760	PUMP-FUEL	016385	KAXTT	393		
3043937-06	EXCITER-IGNITION	162828	KAXTT	362		
3045038-03	VALVE-COMPRESSOR BLEED	938B				
3049382CL03	VANE RING-POWER TURBINE ASSY	HLA4ATAC				
3054917-02	SENSOR-TEMPERATURE, OIL & FUEL	NR00719				
3055000-01	SHAFT-STUB COMPRESSOR ROTOR	PKAAA001219				
3055640-01	SHAFT-STUB COMPRESSOR, FRONT	RWA617093				
3055736-01	TORQUE LIMITER	FBD361				
3055781CL02	VANE RING-TURBINE ASSY, DF	HLA231B6767	CPWUB			
3073900-01	BEARING-RLR,FLG 1.5748X2.6172X.5906	AEB0300235	FAARE			
3073901-01	BEARING-BALL,FLG 1.610X2.834X.745	AEB0300253				
3076154-01	GOVERNOR-PROPELLER	A58029533				
3076290-01	BLADE-POWER TURBINE	215480541				
		YWB837				
		YWB857				
		YWA725				
		YWB839				
		YWB856				
		YWB882				
		YWB894				
		YWB899				
		YWBFB2				
		YWBFB4				
		YWBFB9				
		YWBG24				
		YWBG25				
		YWBG36				
		YWBH6				
		YWBH73				
		YWBH77				

PRATT & WHITNEY CANADA CORP. P68 PROGRAM 4.4.50

PT6A-140 S/N: PCE-VA0567 FINAL ACCEPTANCE TEST RECORD

2019 JAN 14 TEST CELL : 3202 BUILD SPEC.: 1294

ENGINE TESTED AND ACCEPTED IN ACCORDANCE WITH E&TI : 800

FINAL VANE FLOW AREAS FIRST STAGE : 29501 2S2 6.00
SECOND STAGE : 29501 3S3 16.07

PERFORMANCE DATA

TAKE-OFF

SPEC ACTUAL

PROP SPEED		1900.	1900.
S.L.S. STD. DAY POWER		867.	867.
I.T.T. (T5A TRIMMED)	DEG R	1838.	1830.
T5D (DERIVED)	DEG R	1888.	1871.
GAS GEN. SPEED	RPM	36735	36400.
SFC @ 18400 BTU/LB	LB/SHP.HR	0.618	0.595
ENGINE TORQUE	PSI		40.17

PRESSURE RATIO @ 36000 RPM 7.38 7.27

FUEL TYPE : CPW 204

TEST LHV : 18497. BTU/LB

S.G. : 0.822 @ 52. DEG F

OIL TYPE : PWA 521 TYPE II

OIL CONSUMPTION : 0.0 LB/HR

T.O. OIL PRESSURE : 90.8 PSI

OIL TEMPERATURE : 155.9 DEG F

I.T.T. TRIM DELTA T (UNTRIMMED-TRIMMED) : 117.0 DEG F

I.T.T. TRIM DELTA T (UNTRIMMED-TRIMMED) : 65.0 DEG C

I.T.T. TRIM CLASS : 25

TRIM RESISTANCE : 8.90 OHMS

COLD HARNESS RESISTANCE : 0.72 OHMS

HANDLING AND CONTROL SETTINGS

TRIMMED MAX NG : N/A RPM

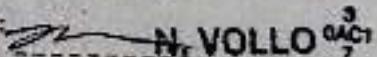
UNTRIMMED MAX NG : 38850 RPM

IDLE SPEED : 20861 RPM

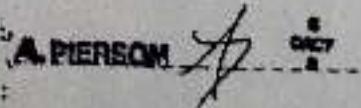
ACCEL. TIME F.I. TO MAX @ T1 : 2.72 SECS @ 46.9 DEG F

ENGINE DRY WEIGHT : 393.48 LBS.

PRODUCTION SIGNATURE



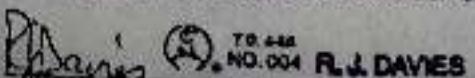
INSPECTION SIGNATURE



GOVERNMENT INSPECTOR :

THE UNDERSIGNED CERTIFIES THAT THIS RECORD ACCURATELY SETS FORTH THE EVENTS DURING THE TEST MADE ON THE ENGINE THEREIN IDENTIFIED.

DATE 15 JAN 2019 FOREMAN, ASSY. & TEST INSP :


TO 44 NO. 004 R.J. DAVIES